







2nd week - Advanced topics 14. July - Evaluating and checking test data interpretation of typical results



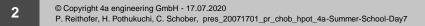
15. July - general yield surface (*MAT_187) and other material models, failure approaches and comprehensive Autofit setup



16. July - Fiber reinforced plastics and their modelling approach an extensive guide



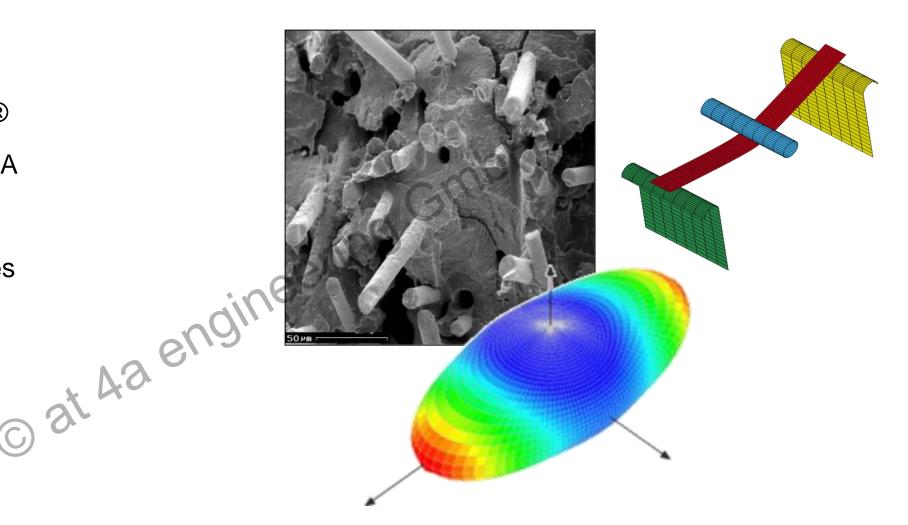
17. July - Python: a powerful tool with VALIMAT®, user defined material cards/specimen



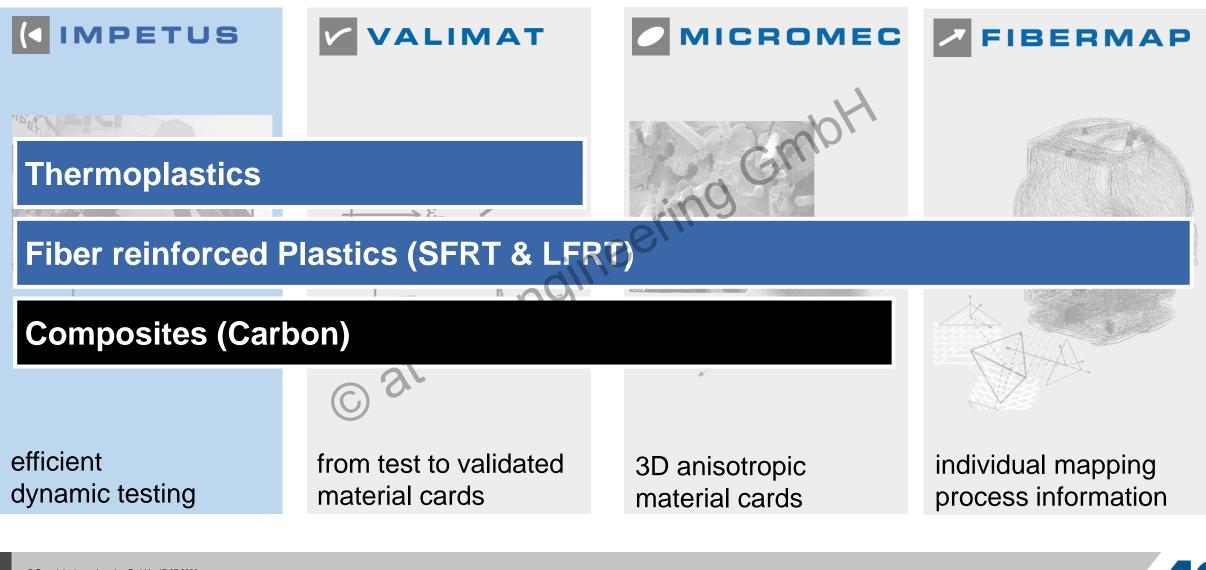
Content

- Recap DAY 6
- Introduction MICROMEC®
- Material model in LS DYNA
- Motivation

- Material model approaches
- Manufacturing influence
- Material characterization
- Casestudy sleeve



intelligent reliable solutions for plastics, composites, metals, foams, ...

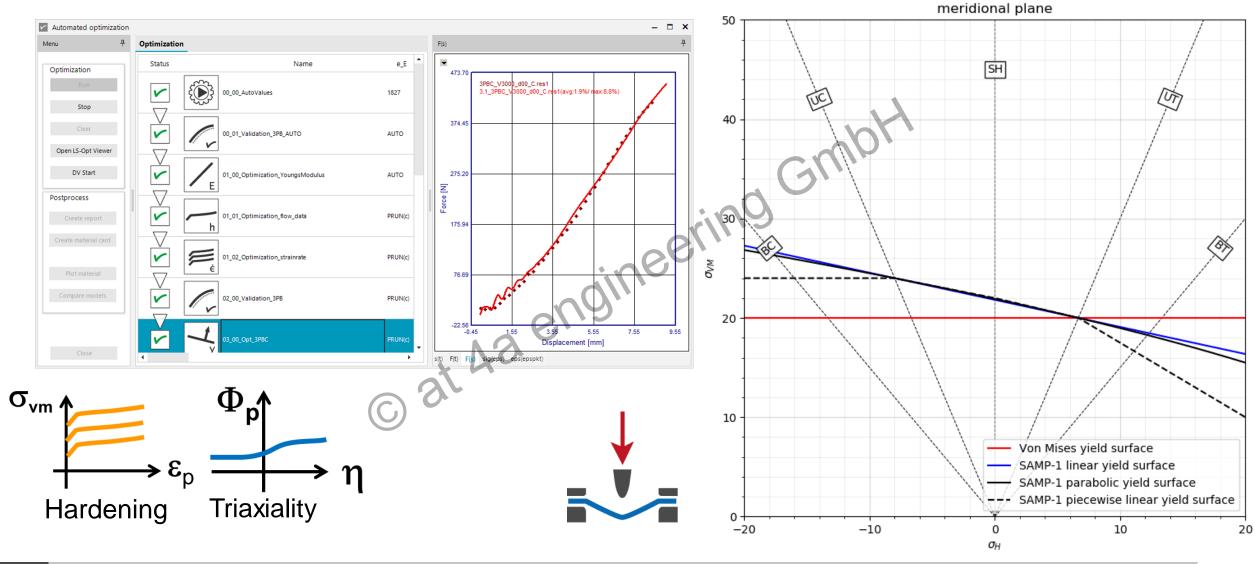


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N P H Y S I C S W E T R U S

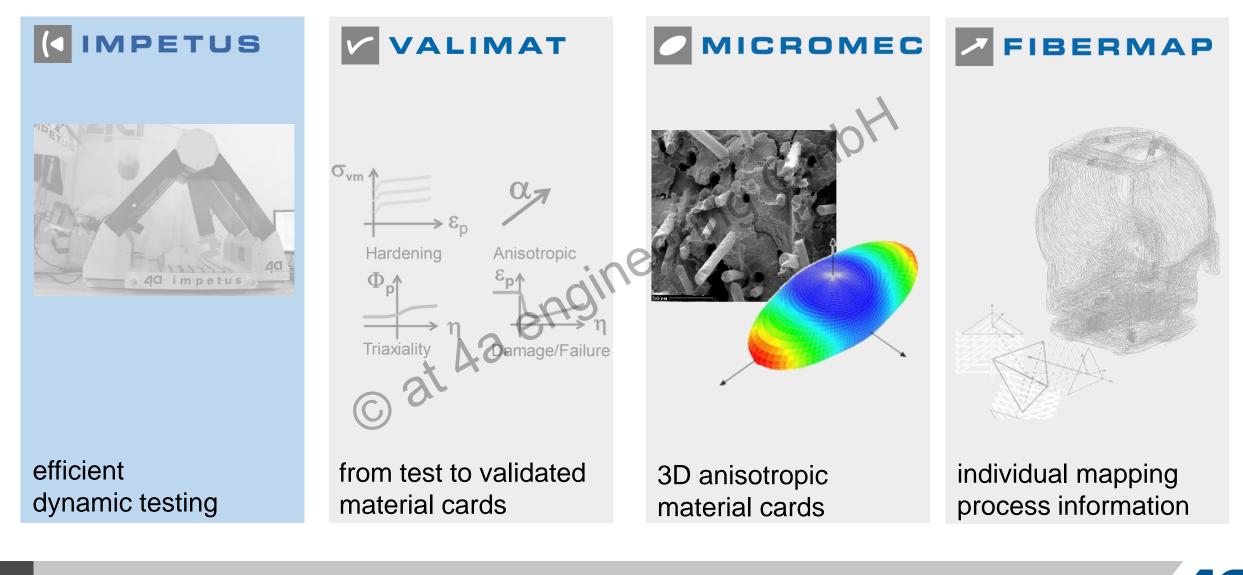
Workflow for Material Card Generation - AUTOFIT







intelligent reliable solutions for plastics, composites, metals, foams, ...

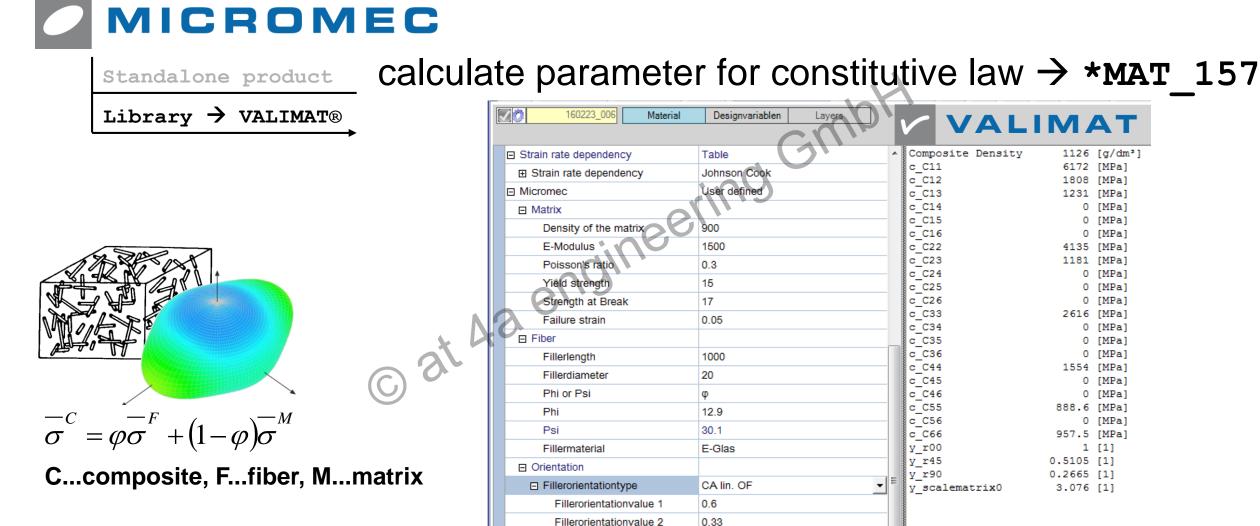


1.

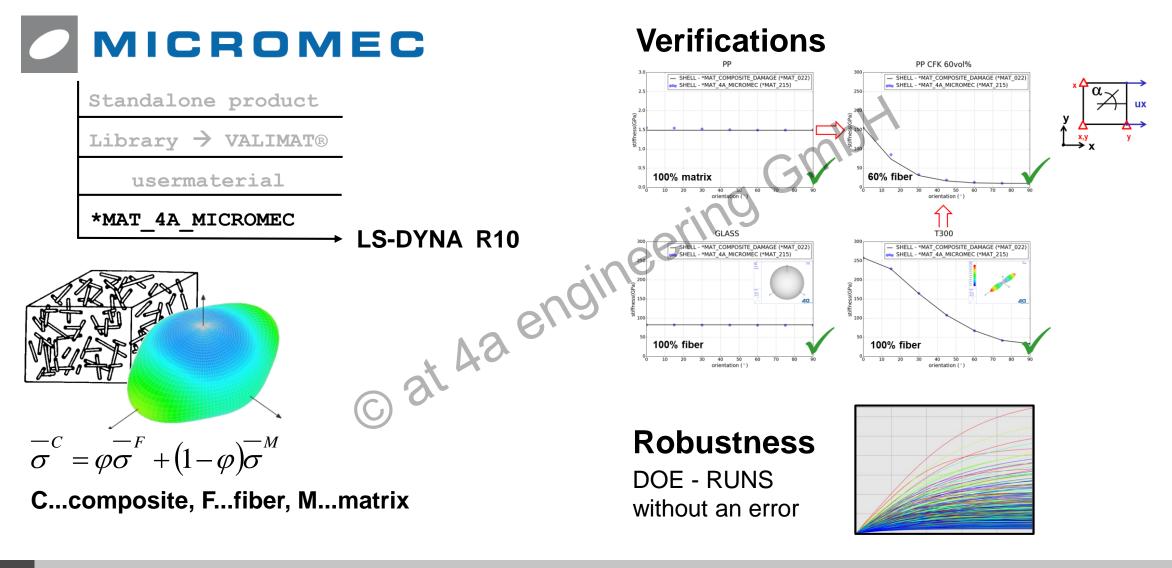
MICROMEC



C...composite, F...fiber, M...matrix



MICROMEC plug able Main Framework possible extensions Mori Tanaka Mean Field Theory Standalone product other plasticity formulations, Library > VALIMAT® $\Delta \varepsilon^{C} \Longrightarrow \Delta \varepsilon^{M}, (\Delta \varepsilon^{F})$ Composite stresses, strains (i) usermaterial $\Delta \varepsilon^{M} = \frac{1}{\varphi \overline{B_{i}} + (1 - \varphi)I} \Delta \varepsilon^{C}$ Matrix stresses, strains \rightarrow Yield condition → J2 Plasticity **Isotropic Harding** $\Delta \varepsilon^{\scriptscriptstyle M} \Rightarrow E^{\scriptscriptstyle T}_{\scriptscriptstyle M}, \Delta \varepsilon^{\scriptscriptstyle M}{}_{\scriptscriptstyle pl}, \Delta \sigma^{\scriptscriptstyle M}$ Fiber stresses, strains $\overline{B}_{i+1} = f\left(fo^{(4)}, E_M^T, \frac{l}{d}\right)$ **Table Lookup or Parameter Setup** Failure, Damage Criteria $\overline{A} = S^F \overline{B}_{i+1} C^M$ $\Delta \sigma^{C} = \left[\varphi \overline{A} + (1 - \varphi) I \right] \Delta \sigma^{M}$ Composite stresses, strains (i+1) $\overline{\sigma}^{C} = \varphi \overline{\sigma}^{F} + (1 - \varphi) \overline{\sigma}^{M}$ fiber C...composite, F...fiber, M...matrix matrix assumption elliptical inclusion (Eshelby Tensor)

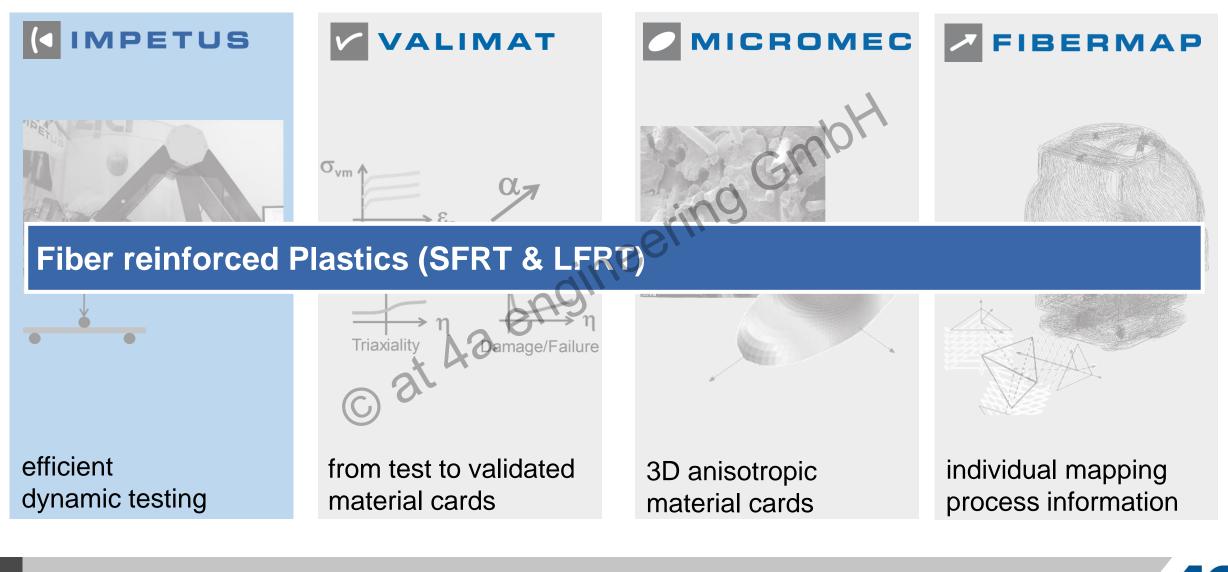




material models in LS DYNA – overview composites

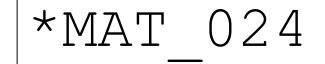
| No. | Elastic | Plastic | Damage | Strain rate | Failure | |
|-------|---------------------|--|------------------------|------------------------|---|---|
| 2 | Ortho / Anisotropic | None | None | None | *MAT_ADD_EROSION | |
| 24 | Isotropic | Mises | None | Plasticity | *MAT_ADD_EROSION | E I |
| 103 | Isotropic | Hill | None | Plasticity | *MAT_ADD_EROSION | SFRT/LFRT |
| 108 | Orthotropic | Hill | None | None | *MAT_ADD_EROSION | =RT/ |
| 157 | Anisotropic | Hill | None | Plasticity | *MAT_ADD_EROSION | SI |
| 215 | *MAT_4a_micromec i | ec in development: Model based on MORITANAKA | | AMEANFIELD | | |
| 22 | Orthotropic | None | None | None | Orientation dependent | |
| 54/55 | Orthotropic | None | Elastic Orthotropic | Strength | Chang-Chang/Tsai-Wu Orientation dependent | lar |
| 58 | Orthotropic | None | Elastic Orthotropic | Strength, Stiffness | mod. Hashin Orientation dependent | Carbon, Glass, Kevlar endless & fabric |
| 158 | Orthotropic | None | Elastic Orthotropic | Visco-elasticity | Orientation dependent | on, Gla dless 8 |
| 261 | Orthotropic | None | Elastic Orthotropic | None | failure Pinho (Puck) Orientation dependent | Carbo en |
| 262 | Orthotropic | None | Elastic Orthotropic | None | failure Camanho (Puck) Orientation dependent | |

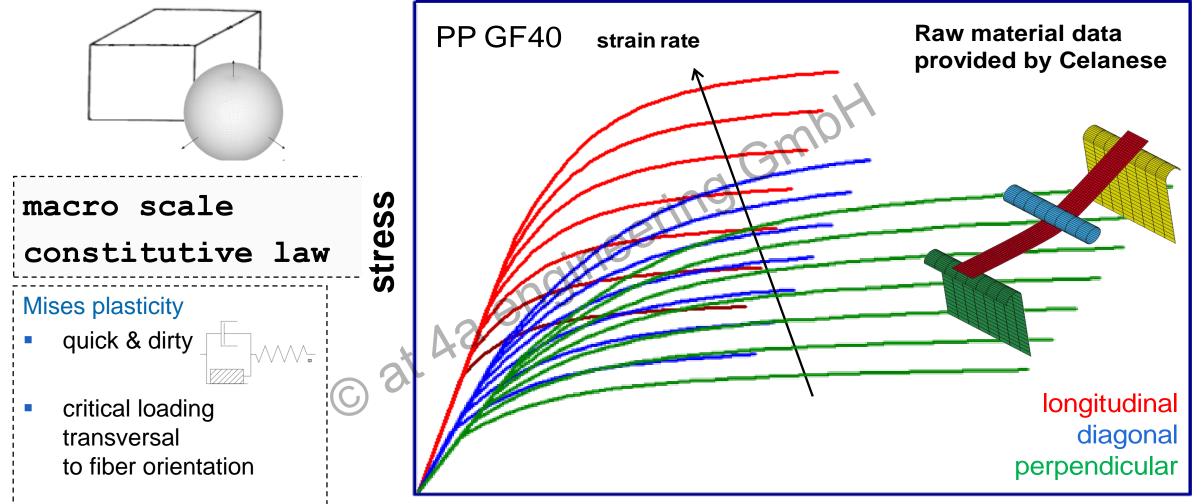
Intelligent reliable solutions for plastics, composites, metals, foams, ...



1.

Motivation – current simulation standard

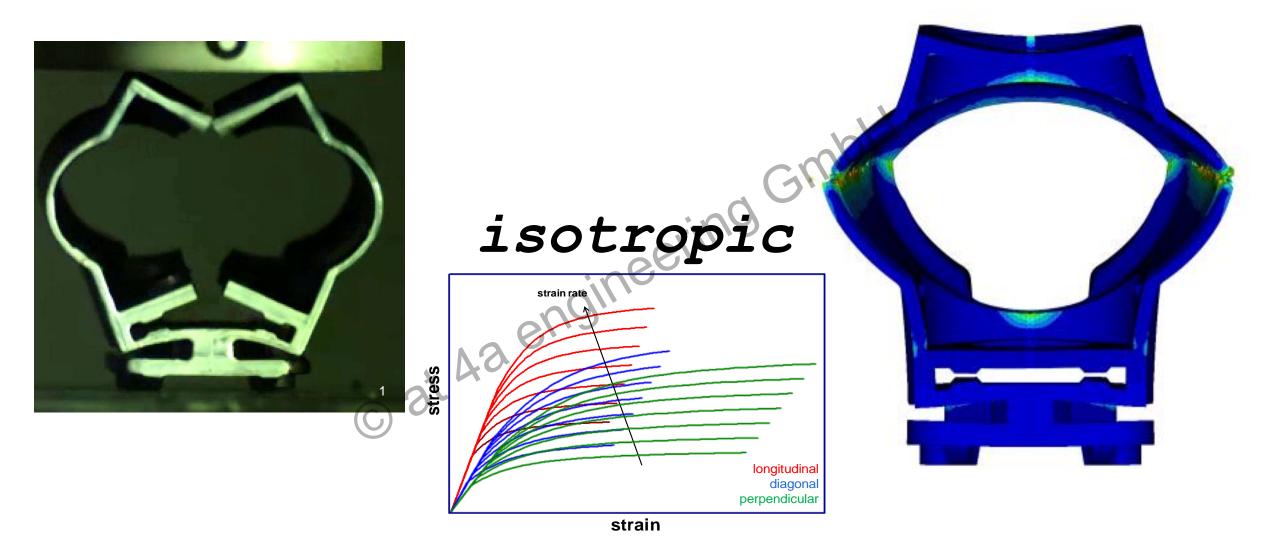




strain



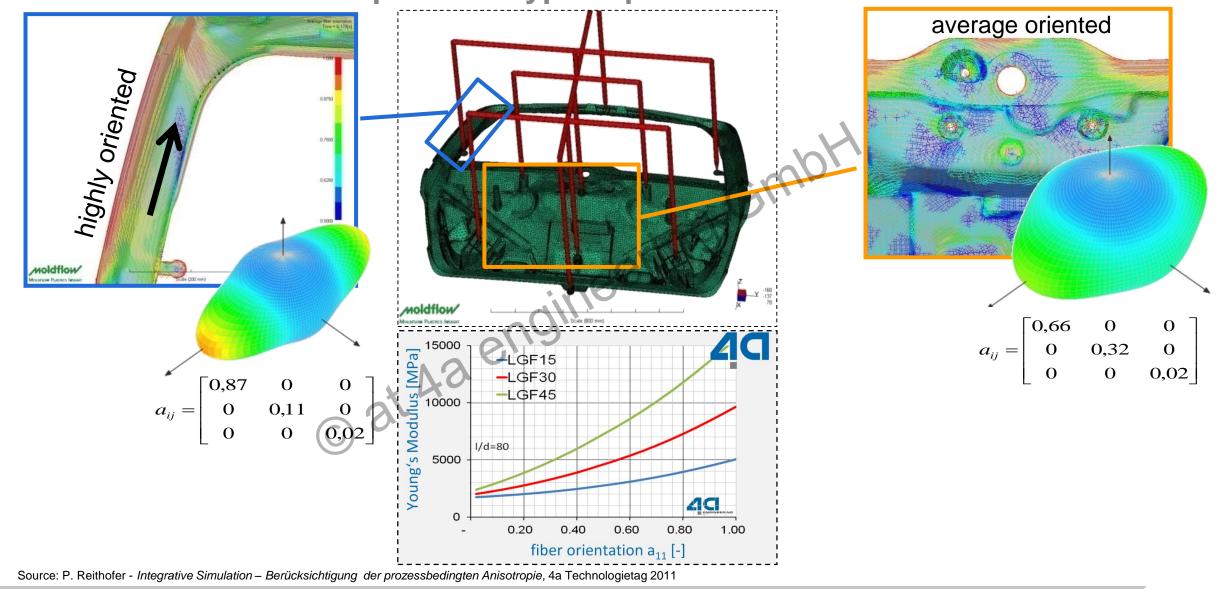
Motivation – current simulation standard



See more: R. Steinberger, et.al. Hirtenberger Automotive Group – Considering the Local Anisotropy of Short Fiber Reinforced Plastics, European Dynaforum 2017

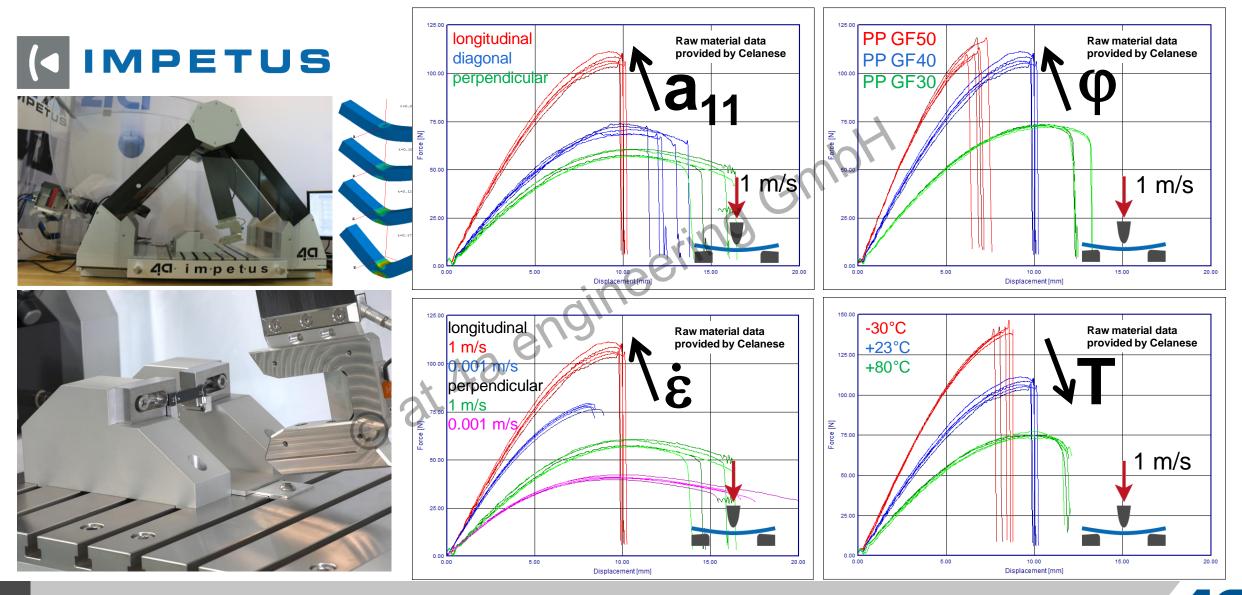
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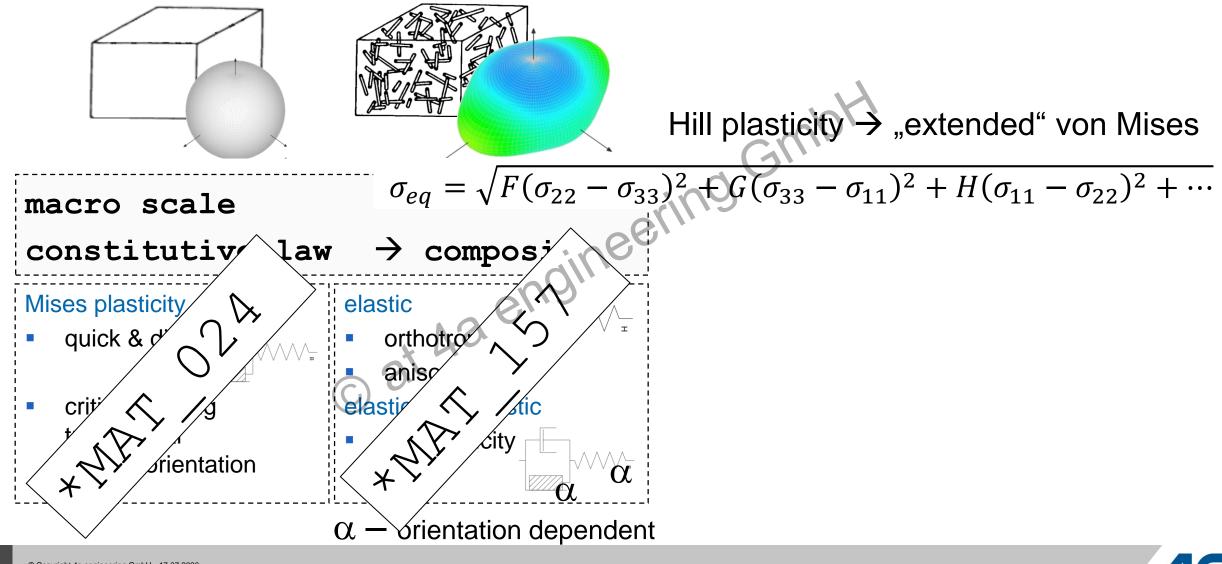
Fiber orientation – development in typical part

Typical material behavior – SFRT / LFRT



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Typical material models in LS-DYNA



material models in LS DYNA – short and long fiber reinforced thermoplastics

1.

| description | variables | Number of variables | dependencies |
|--------------------------|---|---------------------|--|
| anisotropic stiffness | Cij | 21 | $C_{ij}(a_{ij},\varphi,C^M,C^F)$ |
| Hill plasticity | 3D: F, G, H, L, M, N 2D: r00, r45, r90 | 6 | $f(a_{ij}, \varphi, \sigma^M, \sigma^F)$ |
| stress-strain curve | Loadcurve | 3 | $f(a_{ij}, \varphi)$ |
| failure | Loadcurve | 68N9. | $f(a_{ij}, \varphi)$ |

- Not possible to generate samples with explicit defined and varying a_{ii}
- Hard to characterize, too many possibilities in a_{ii}

\rightarrow Micro mechanical model is needed

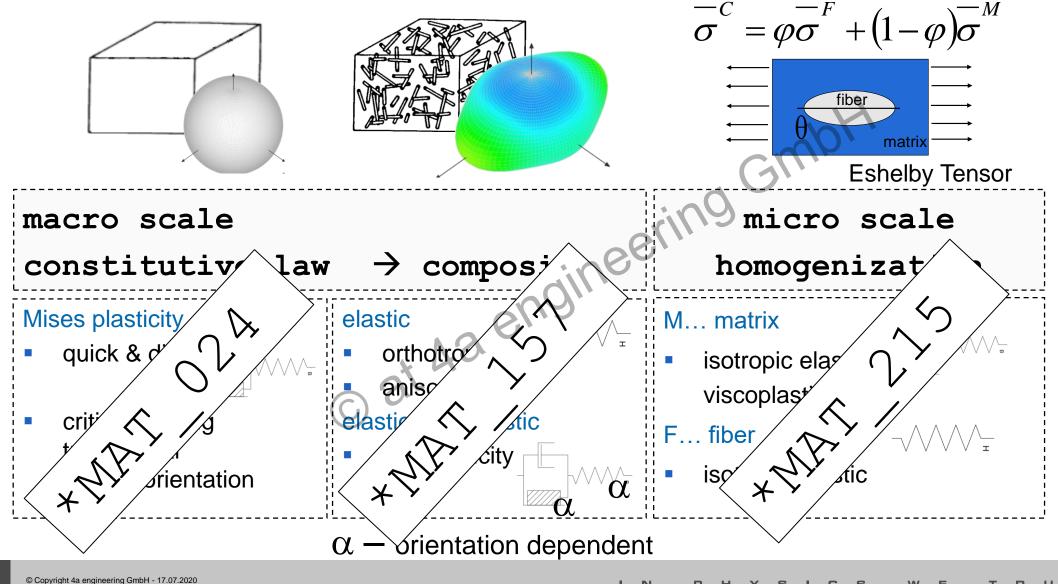
18

a11=0.85 a11=0.7 a11=0.5

a11=1

PA6GF30_08

Typical material models in LS-DYNA



CARD 1: General Options / Parameter

CARD 2-3: Element orientation*

analog to LSDYNA standard anisotropic material cards

CARD 4: Composite Buildup*

| | _ | | | | | | | | |
|--|-------|---|---------|------|-----|------|------|---|--|
| Card 4 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | |
| | FVF | | FL | FD | ine | A11 | A22 | | |
| PP GF30 | -0.3 | | 200.0 | 10.0 | 0, | 0.7 | 0.25 | | |
| PP LGF50 | -0.5 | | 1000.0 | 20.0 | | 0.65 | 0.30 | | |
| PA6 GF45 | -0.45 | | 250.0 | 10.0 | | 0.8 | 0.15 | | |
| Carbon UD | 0.6 | | 10000.0 | 10.0 | | 1.0 | 0.0 | | |
| FVF > 0: fiber volume fraction \rightarrow Composite | | | | | | | | | |
| FVF < 0: fiber mass fraction \rightarrow SFRT/LFRT | | | | | | | | | |

exemplary values without any warranty

20

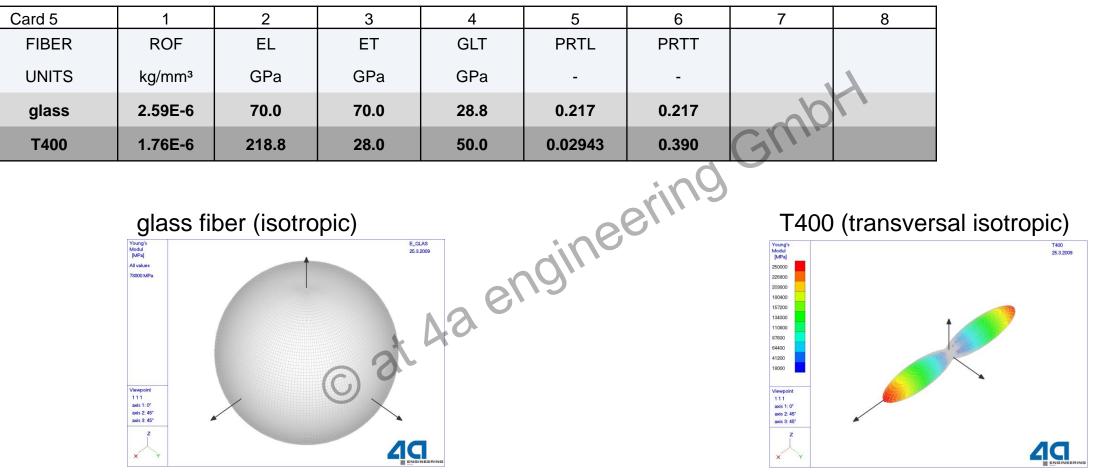
*INITIAL STRESS SHELL/SOLID

*may be overwritten by



US

CARD 5: fiber material



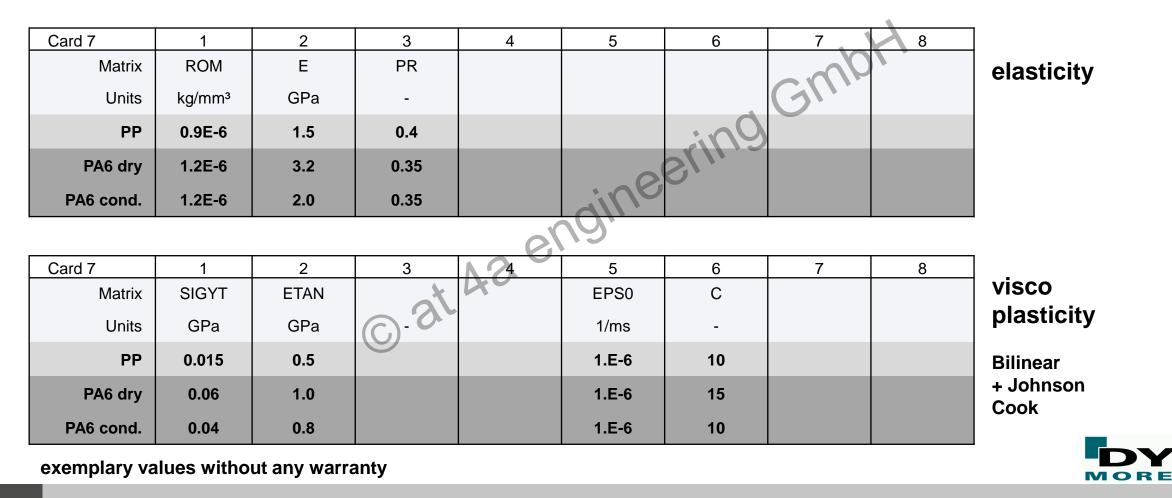
exemplary values without any warranty

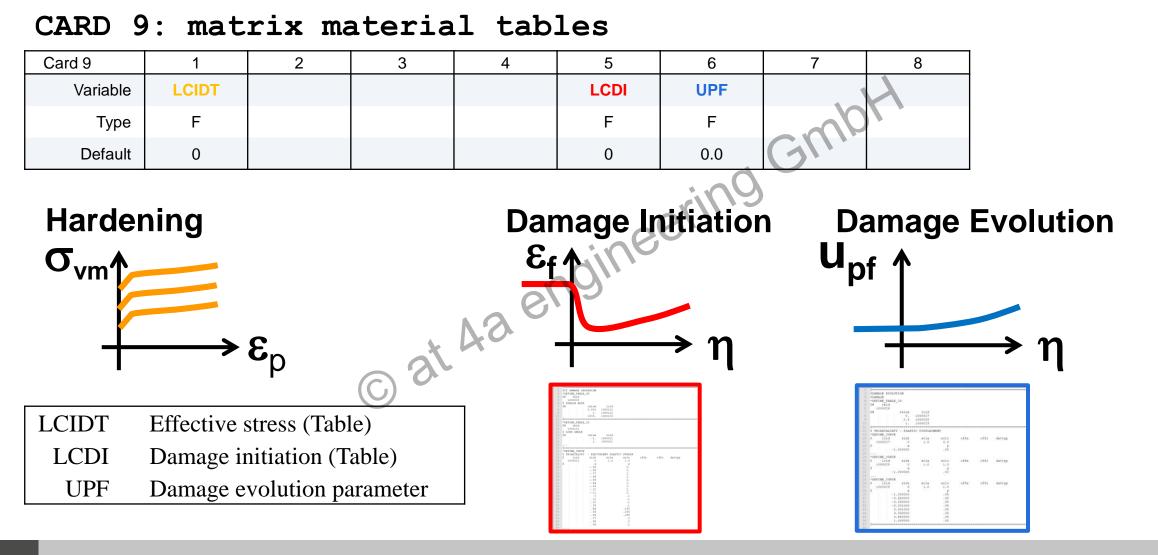
MORE



CARD 7-8: matrix material

from material characterization (e.g. VALIMAT® Workflow)









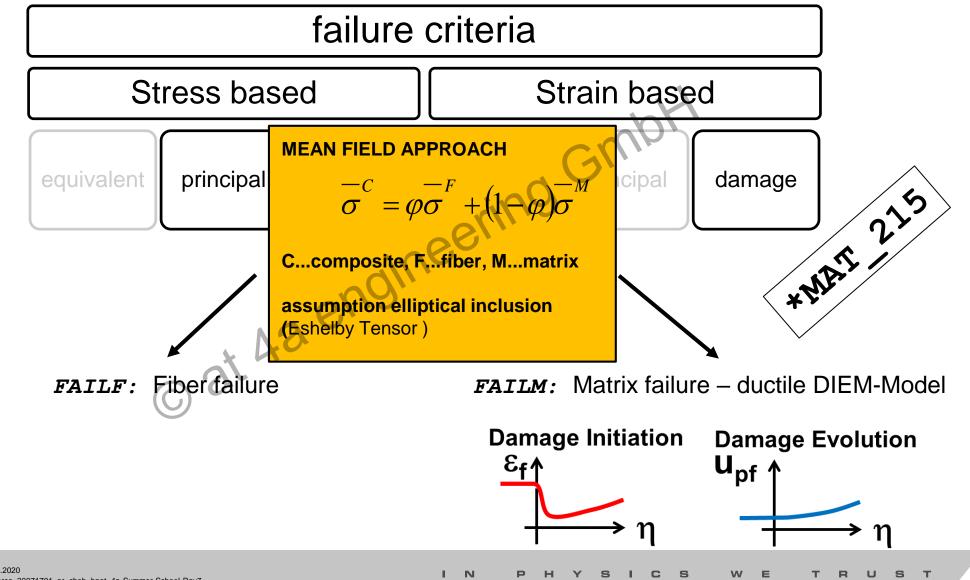
| VariableMIDMMOPTBUPDFAILMFAILMFAILFNUMINTTypeA8FFFFFDefaultnone0.00.010.00.01.0 | Card 1 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | |
|---|----------|------|-------|------|-----|-------|-------|-------|--------|--|
| Default none 0.0 0.0 1.0 | Variable | MID | MMOPT | BUPD | | | FAILM | FAILF | NUMINT | |
| ineering of the second | Туре | A8 | F | F | | | F | F | F | |
| a engineering | Default | none | 0.0 | 0.01 | | | 0.0 | 0.0 | 1.0 | |
| | | | | | aer | ngine | erino | | | |

History#4 (step8: 0-0.81): dm - matrix damage init.

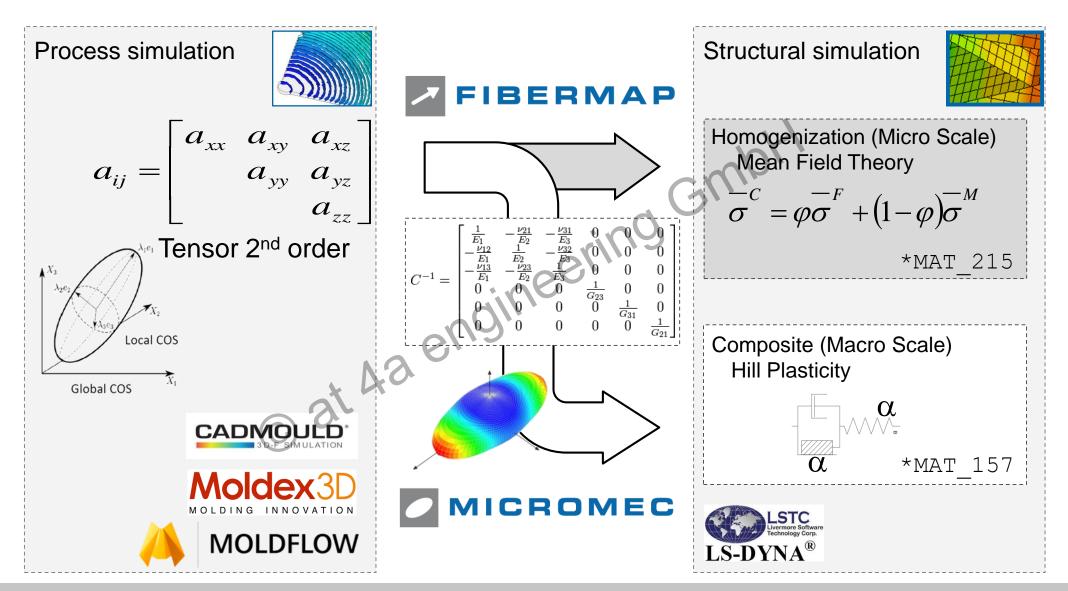
History#6 (step8: 0-0.13): Fiber damage init.



Micro mechanical motivated failure



Material models – present approaches



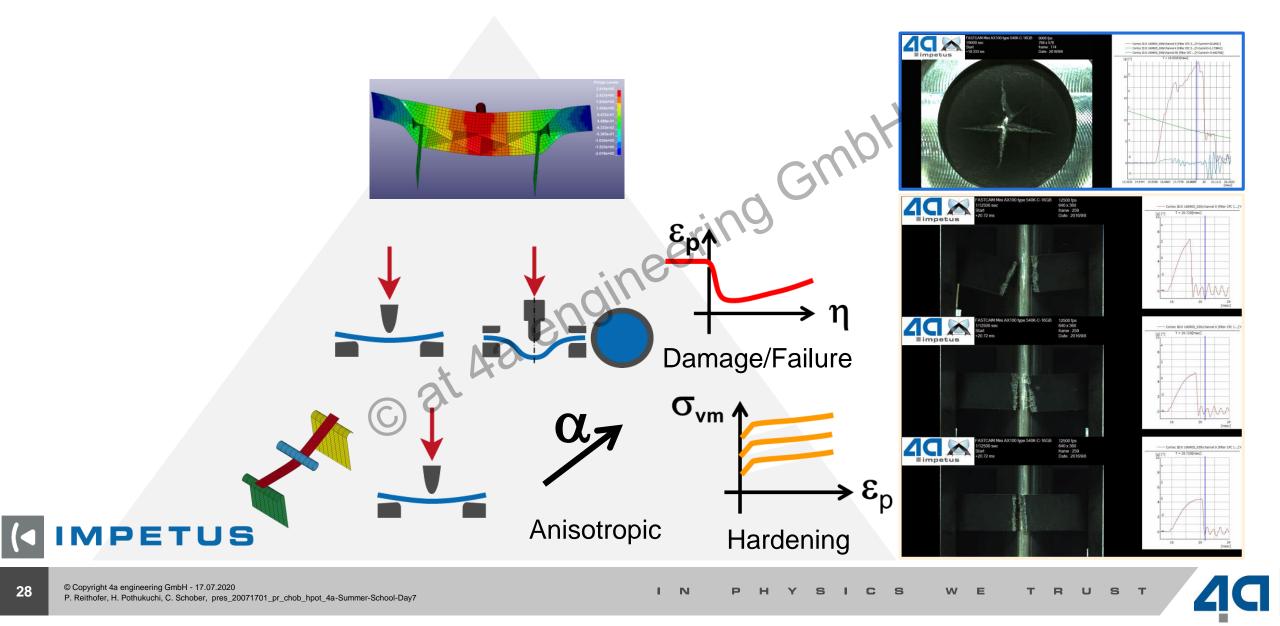


From test to material card

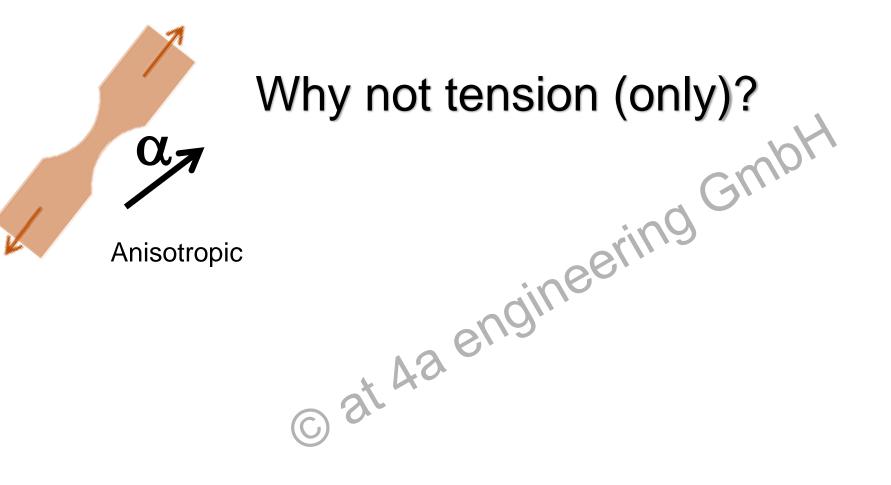


from test to material card





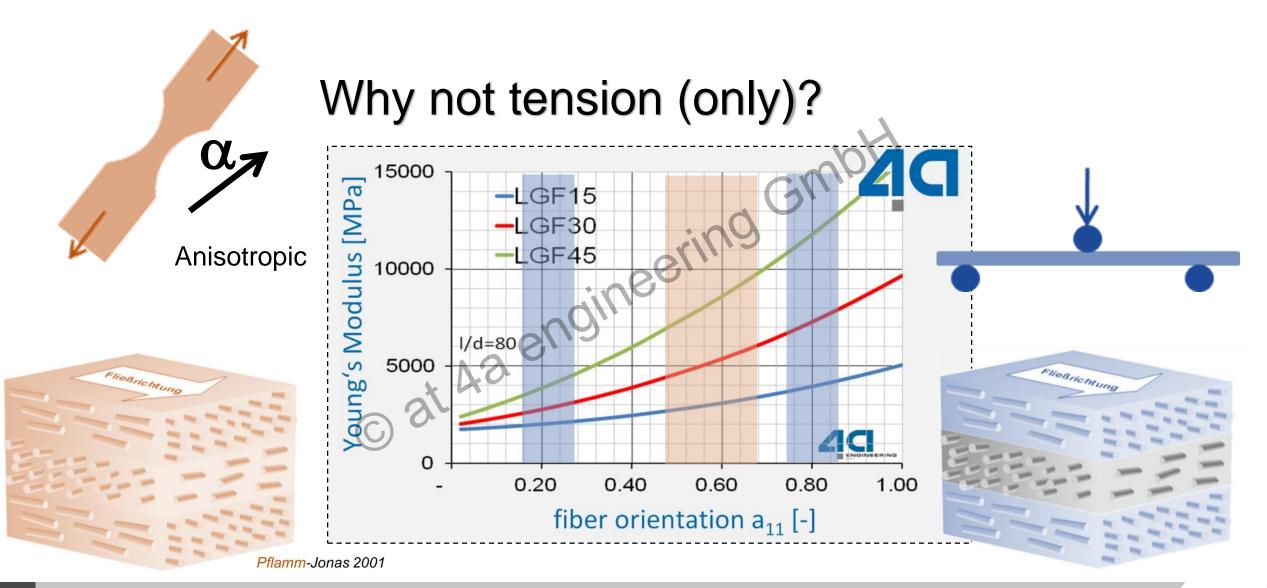
From test to material card



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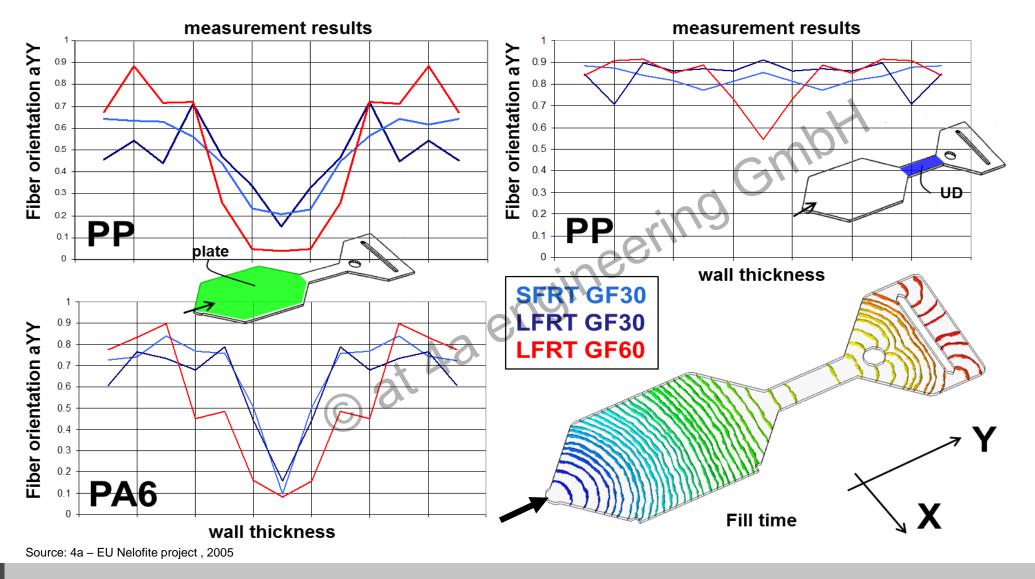


From test to material card

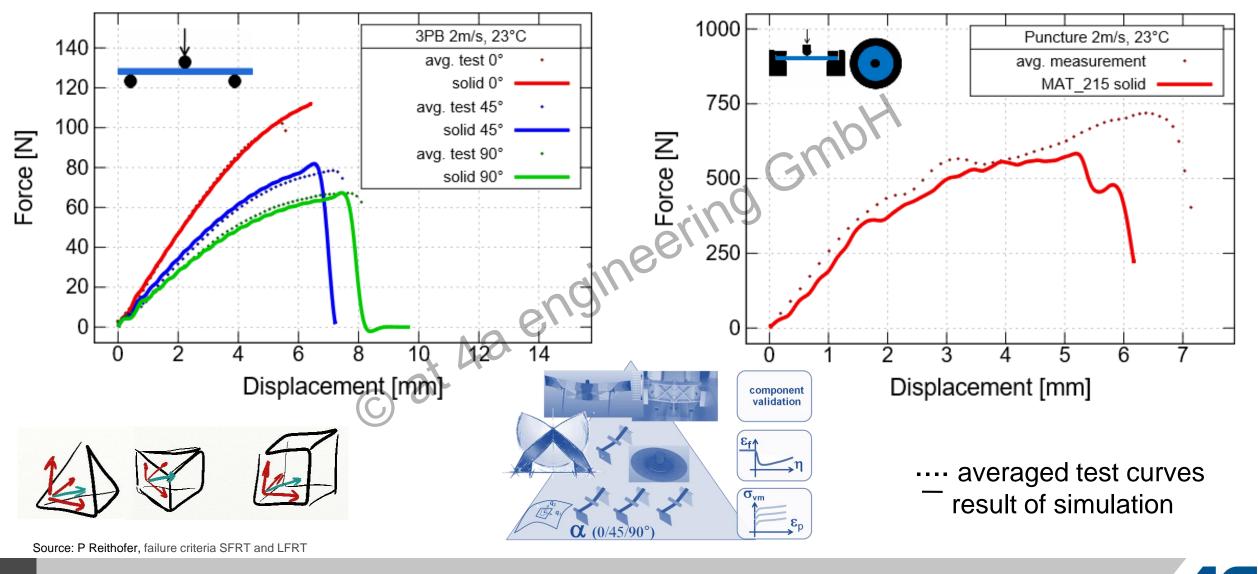


1.

Fiber orientation – development based on flow / viscosity

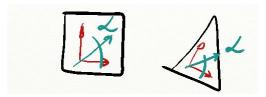


From test to material card – PP LGF30 *MAT 215



From test to material card – PP LGF30 *MAT 215





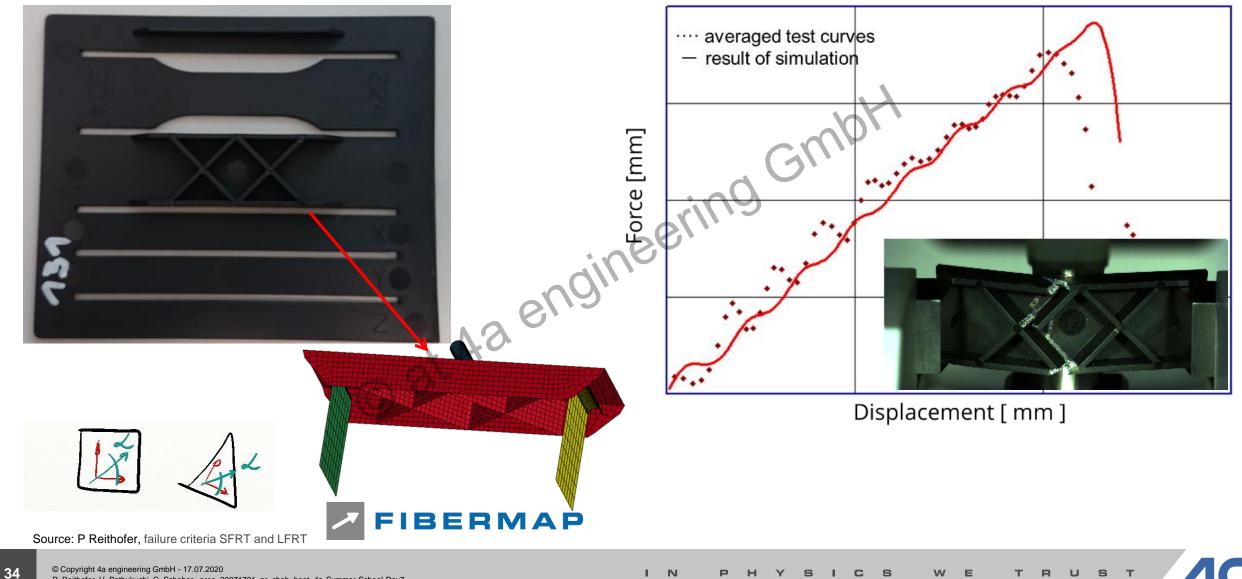
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Source: P Reithofer, failure criteria SFRT and LFRT

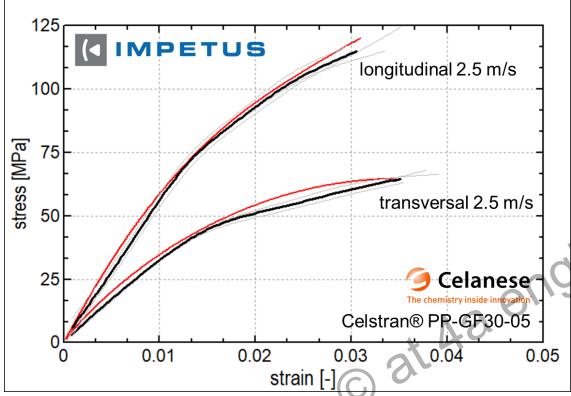
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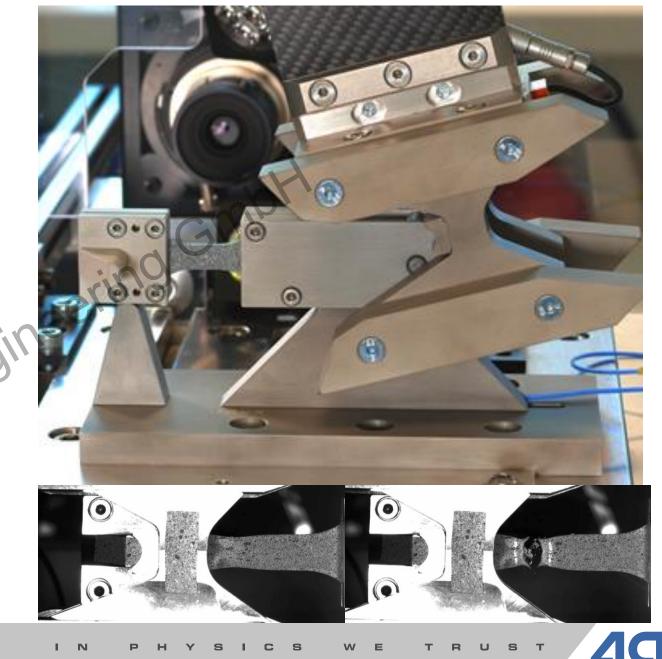
Validation for PP LGF30



Dynamic tensile testing



comparison IMPETUS[™] impact tensile versus classical servo hydraulic test



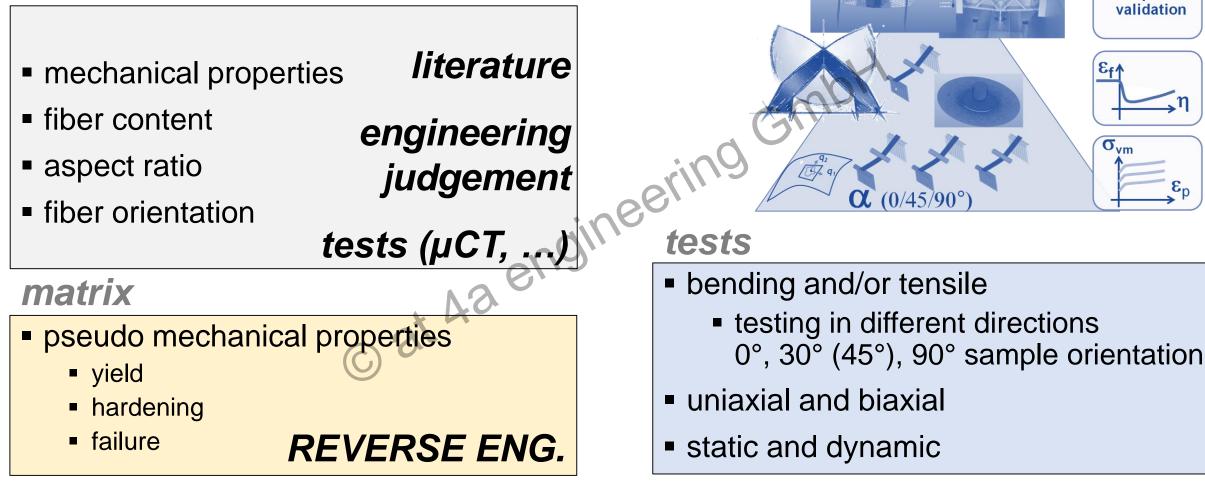


How to get **MAT_157* or **MAT_215*?



How to get *MAT_215?

fiber





component validation

 σ_{vm}

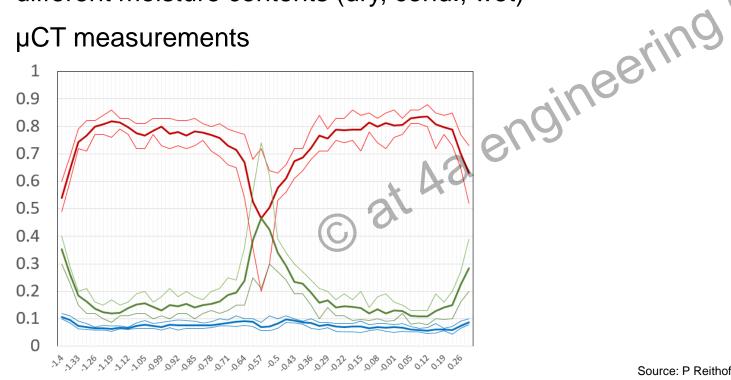
C (0/45/90

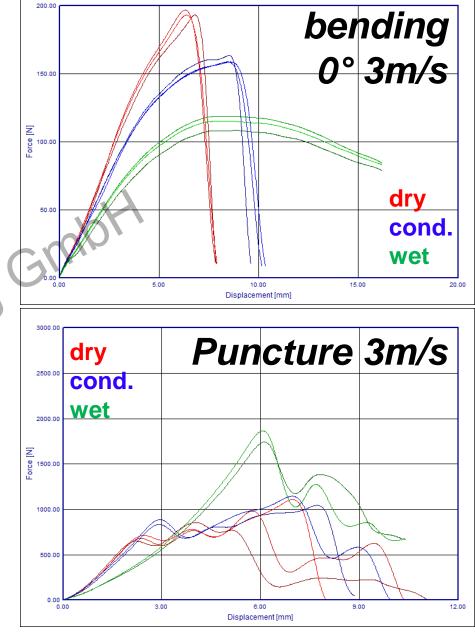


How to get *MAT_215 – case study PA6 GF30 I

Provided by consortium (PCCL, HILTI, Hirtenberger)

- plaques for puncture tests
- bending samples $(0^\circ, ~30^\circ, 90^\circ)$
- different moisture contents (dry, cond., wet)
- µCT measurements





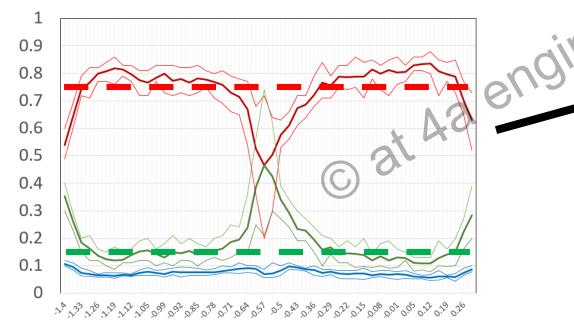
Source: P Reithofer, failure criteria SFRT and LFRT

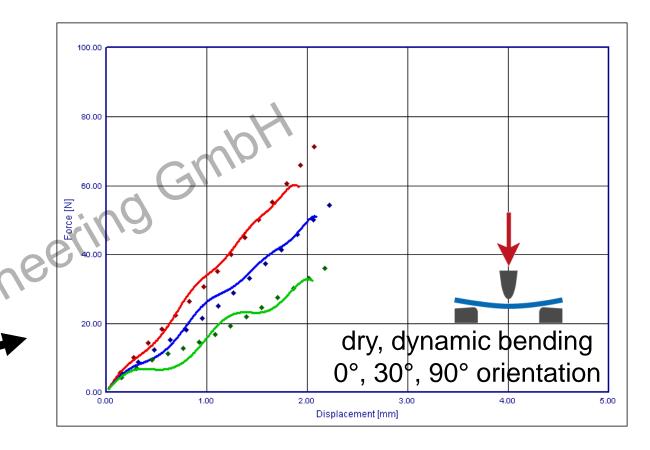
How to get *MAT_215 - case study PA6 GF30 I



1st step: set up the composite

- Fiber properties from literature
- Fiber content 30% wt \rightarrow -0,3
- Aspect ratio typical for short fibers I/d=20
- μ CT measurements \rightarrow average

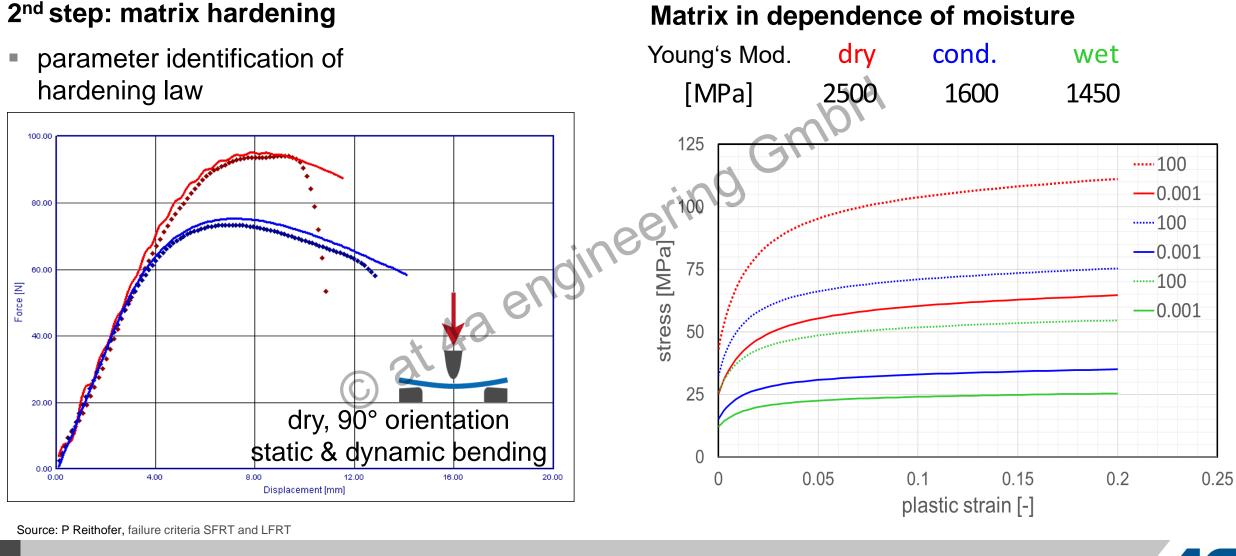




Source: P Reithofer, et.al., failure criteria SFRT and LFRT

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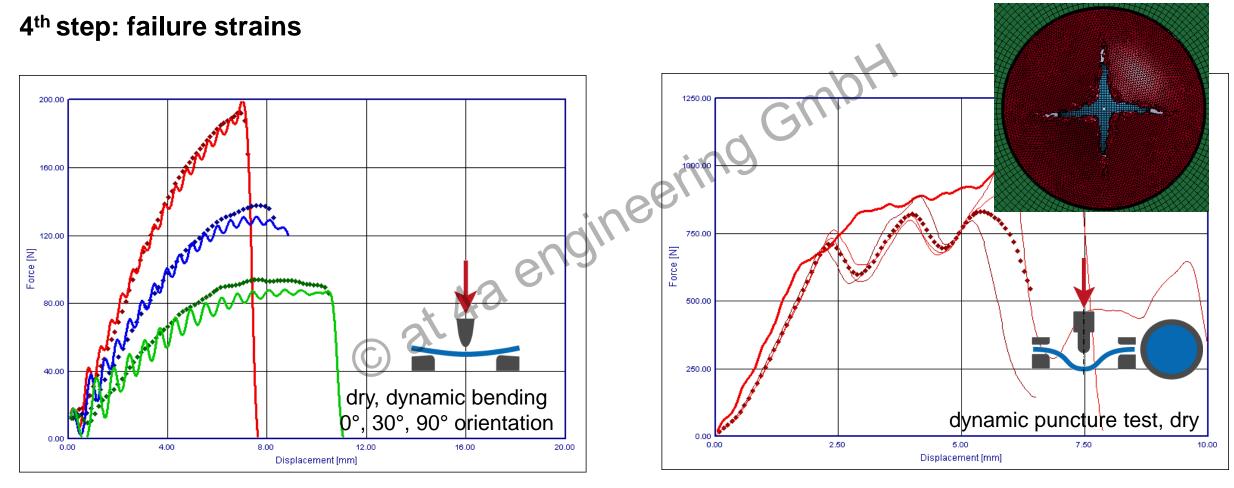




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How to get *MAT_215 - case study PA6 GF30 I

3rd step: validation on dynamic bending

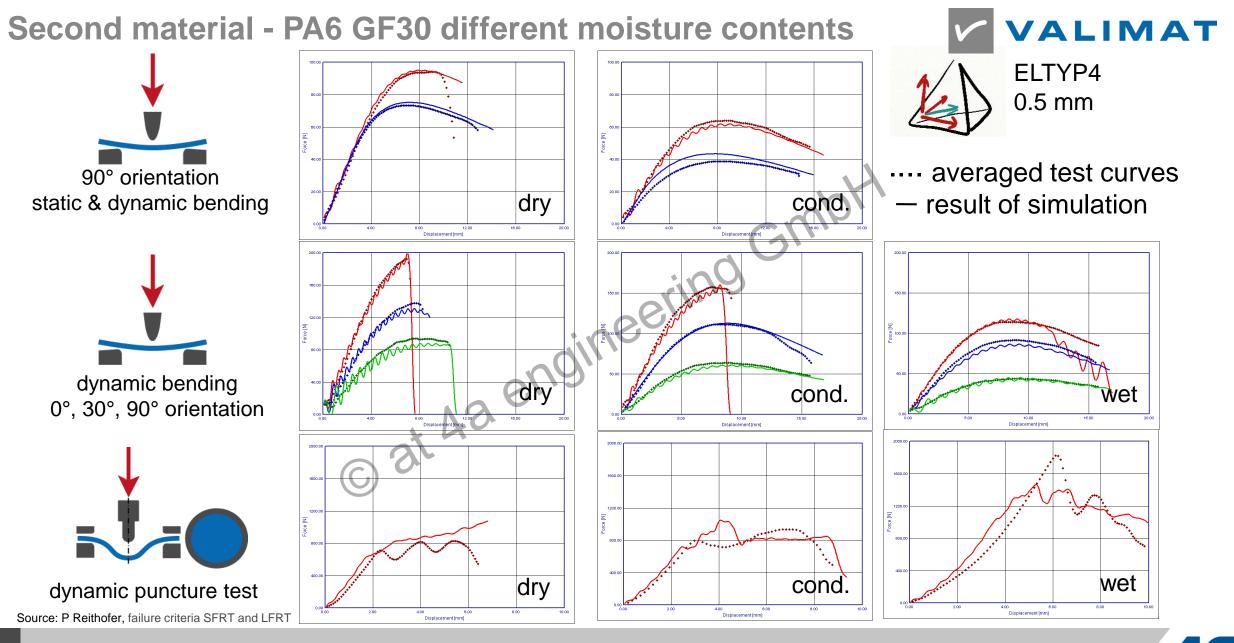


5th step: validation on dynamic puncture

Source: P Reithofer, failure criteria SFRT and LFRT

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What else ?



injection mold for material characterization

DOM & Wall thickness

Melt- & Weldlines



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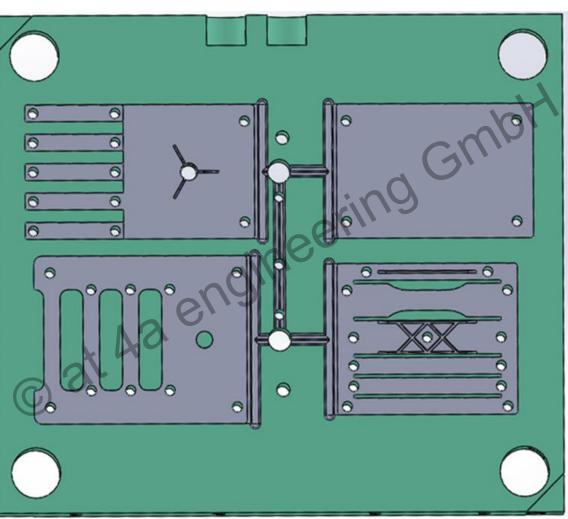


Plate 120 x 80 x 2 mm

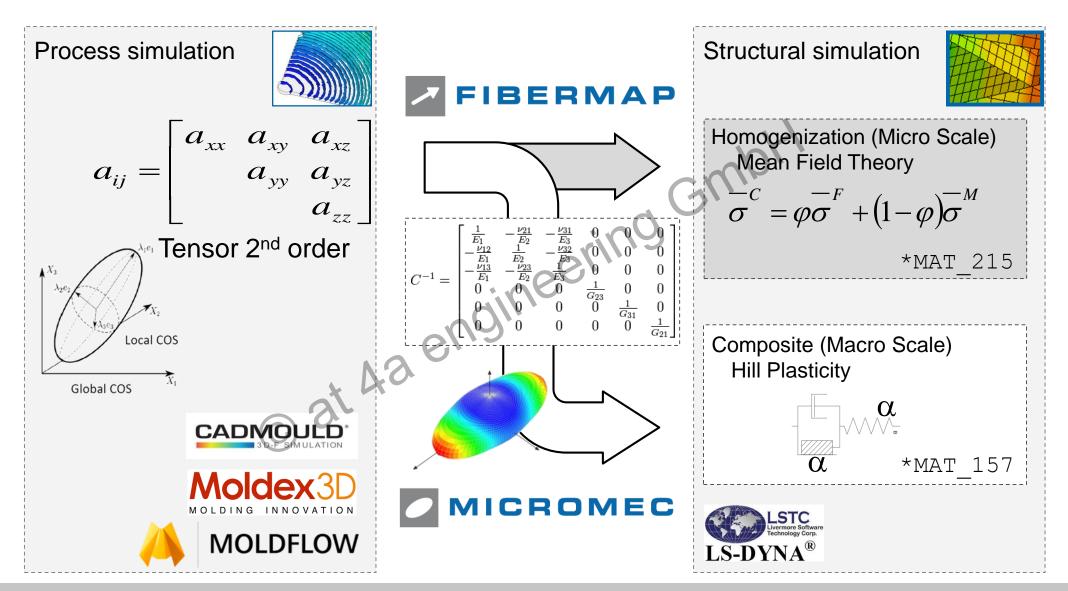


Multi-Specimen & Rib & Component

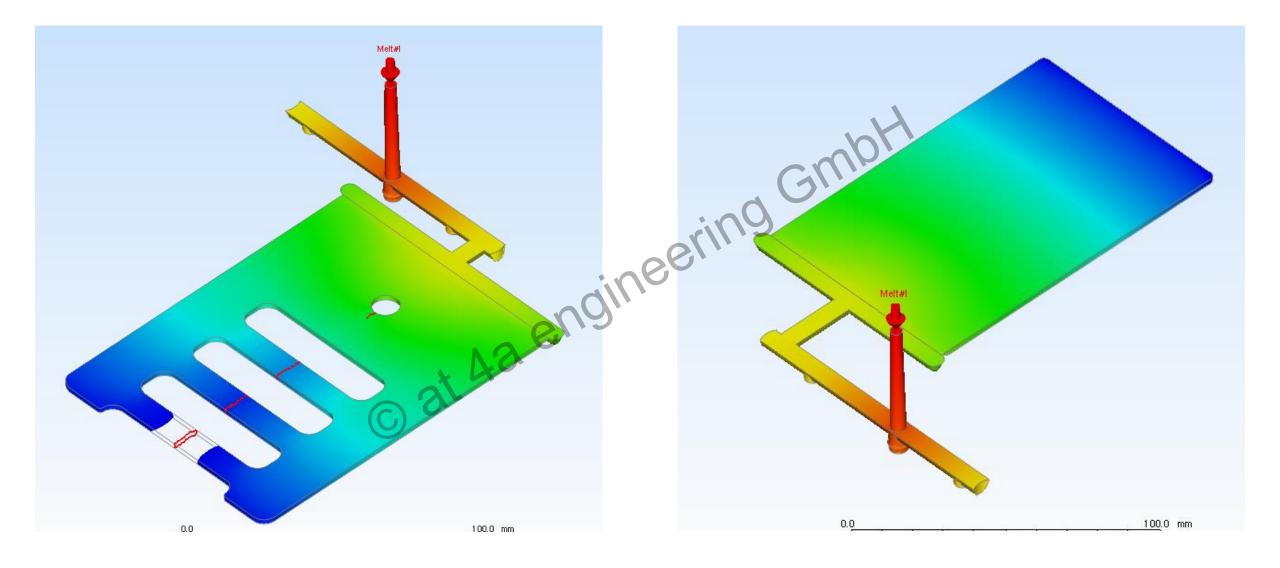


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Material models – present approaches

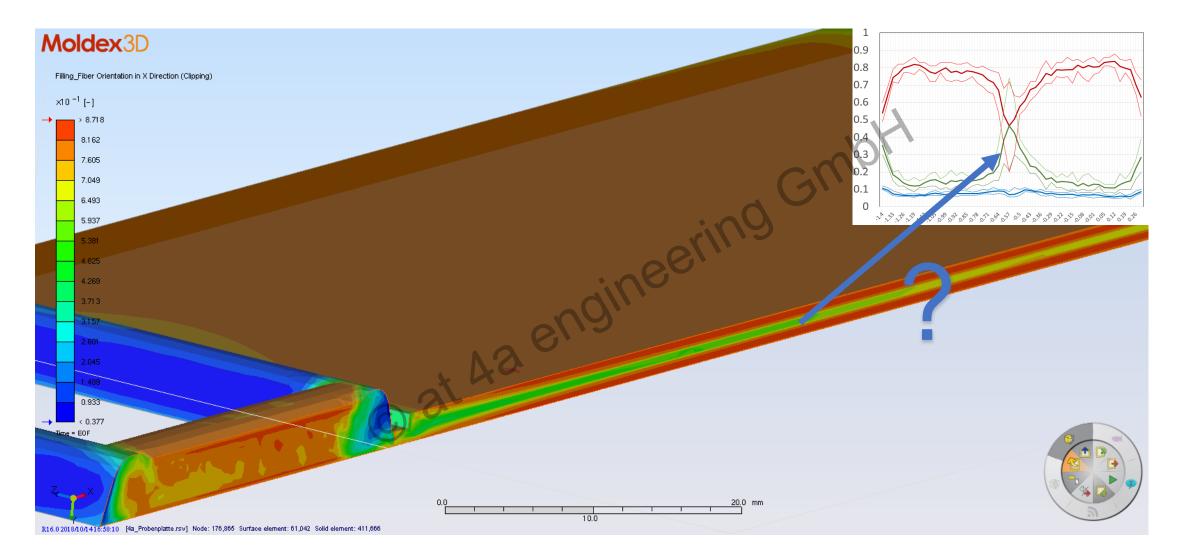


injection mold – process simulation



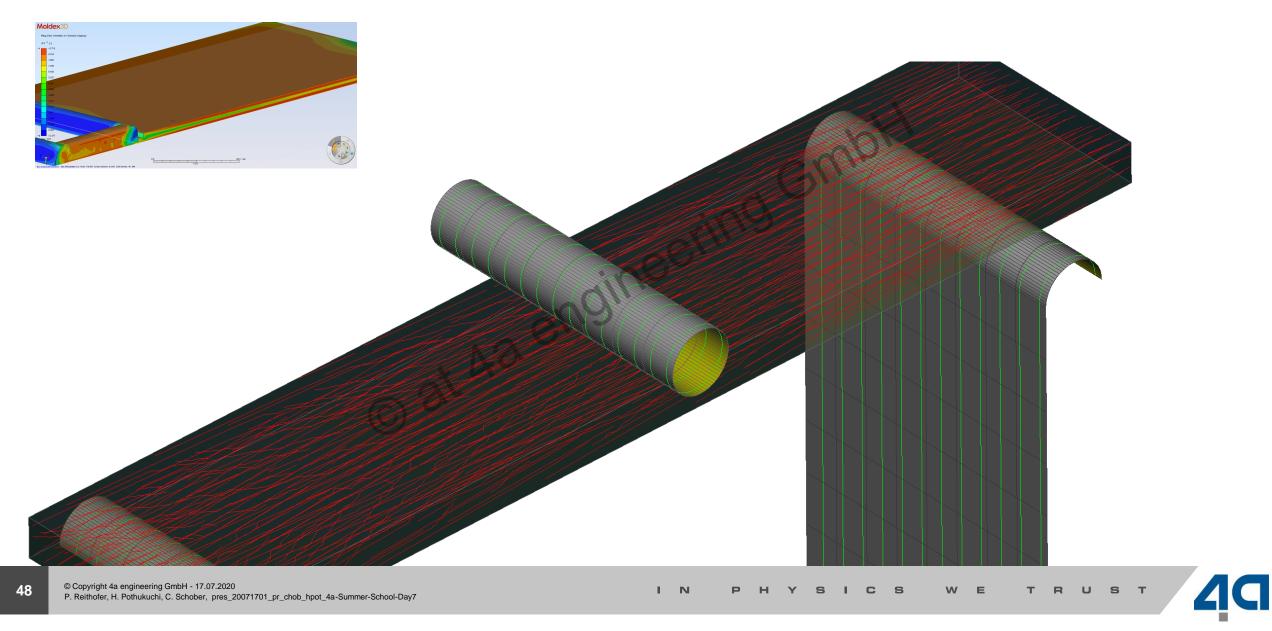
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injection mold – process simulation fiber orientation

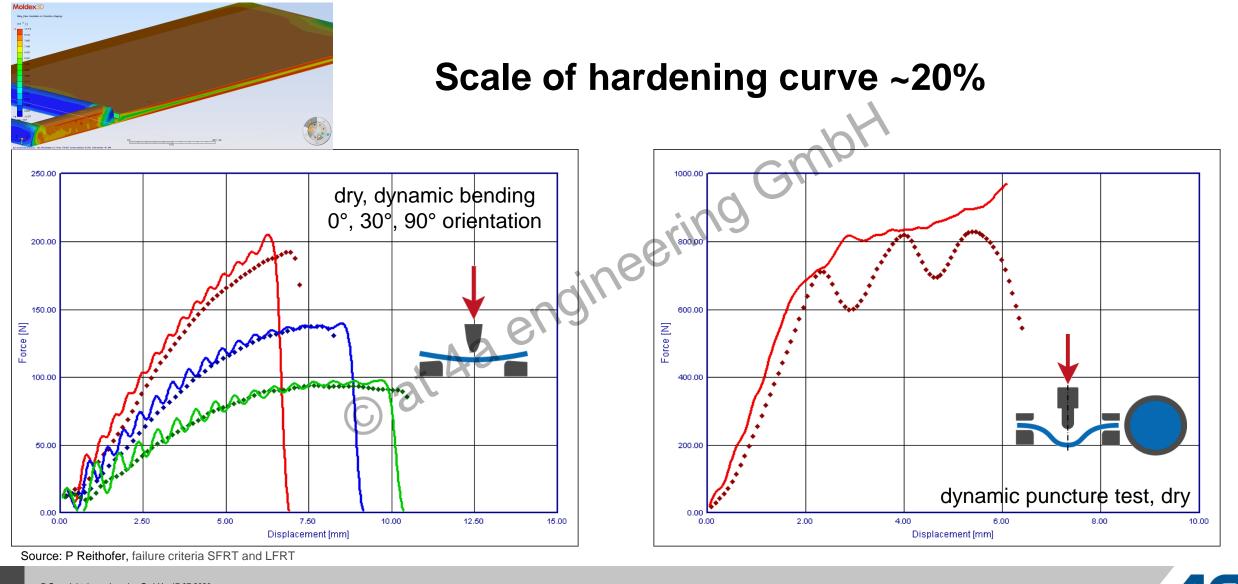






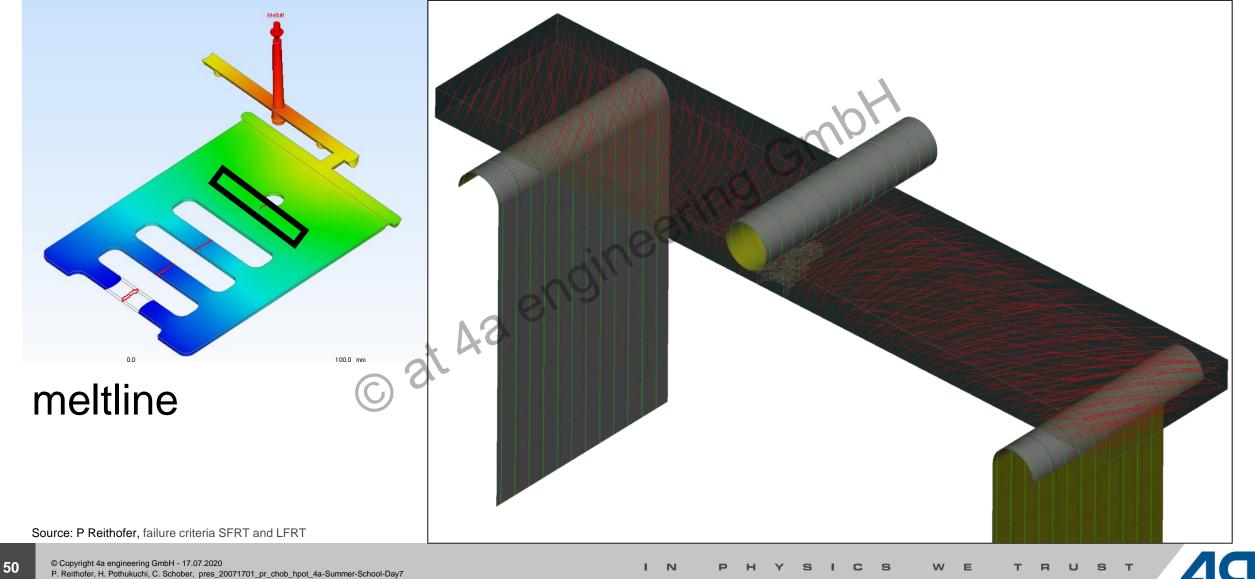




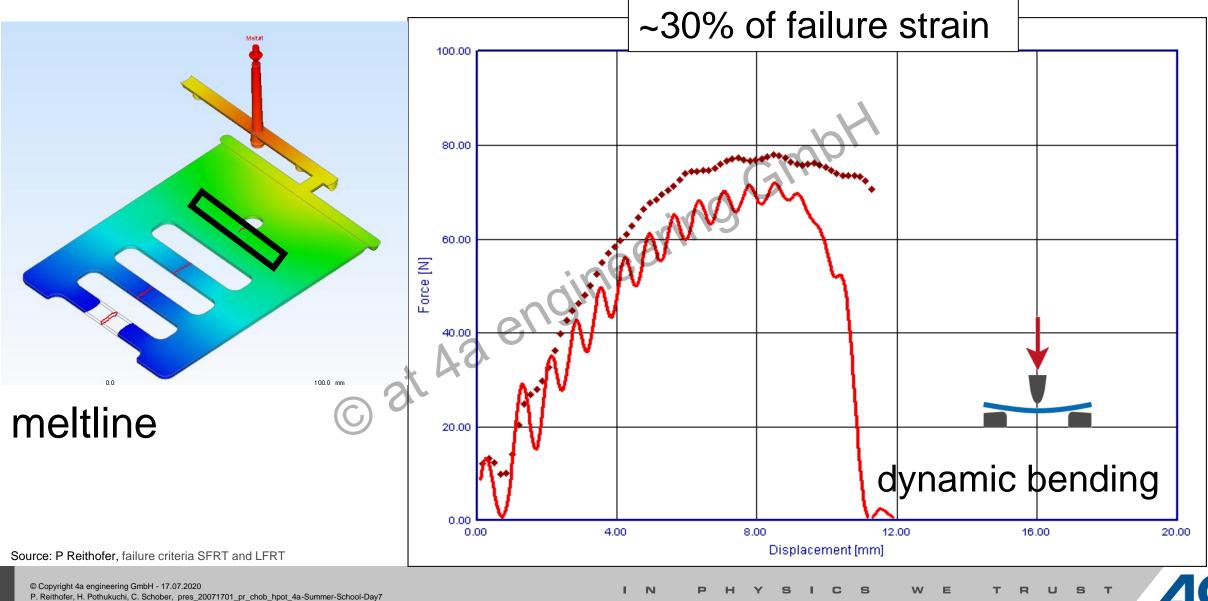


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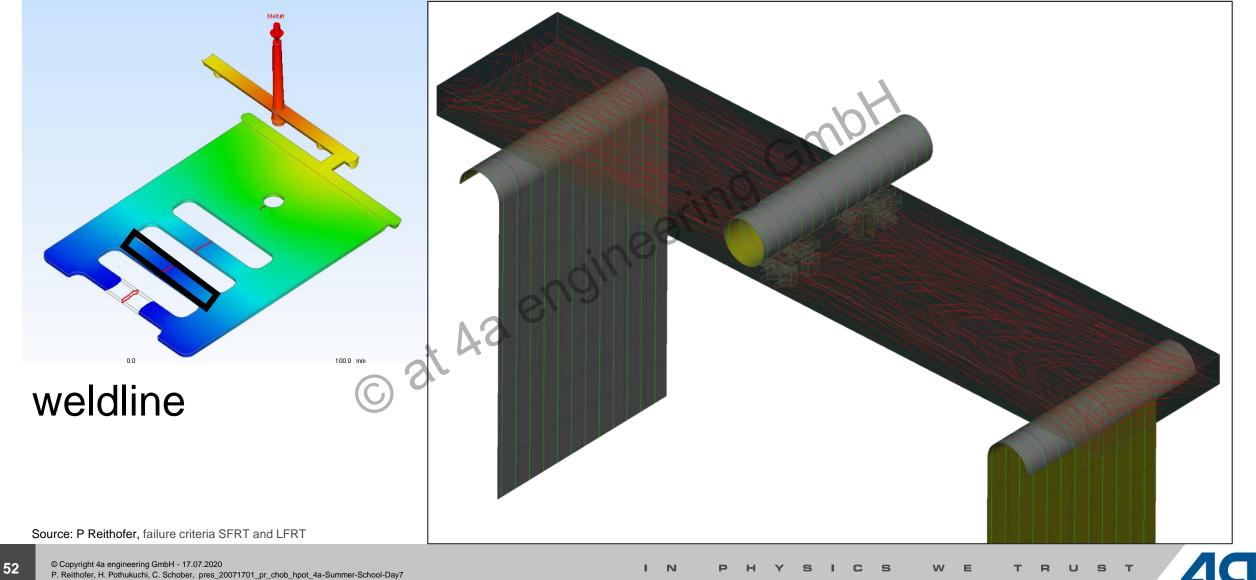




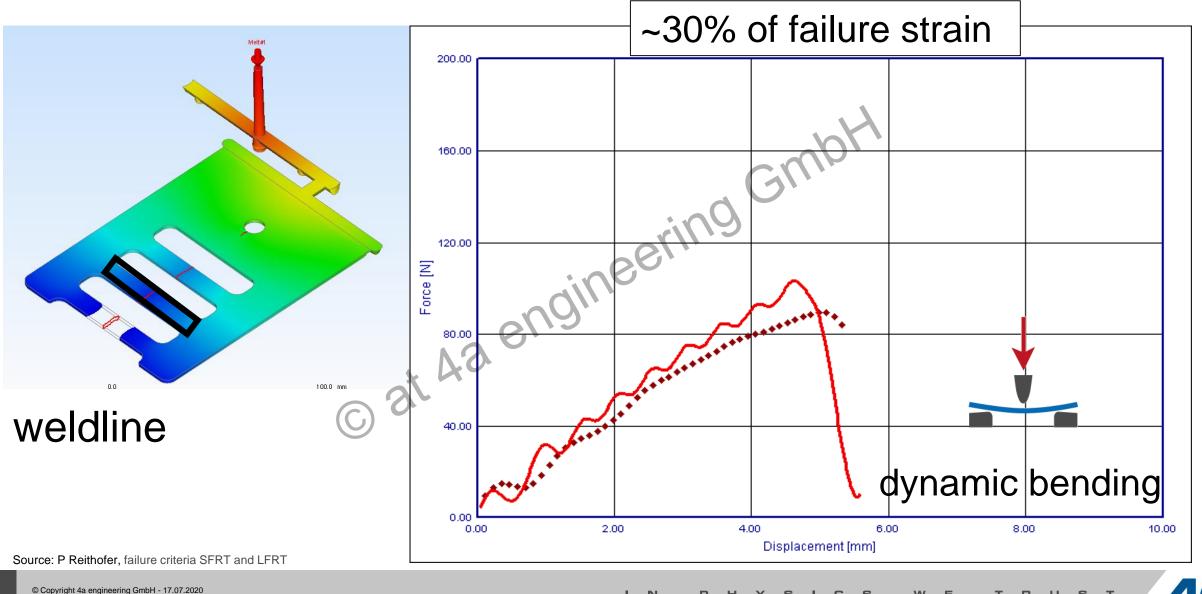












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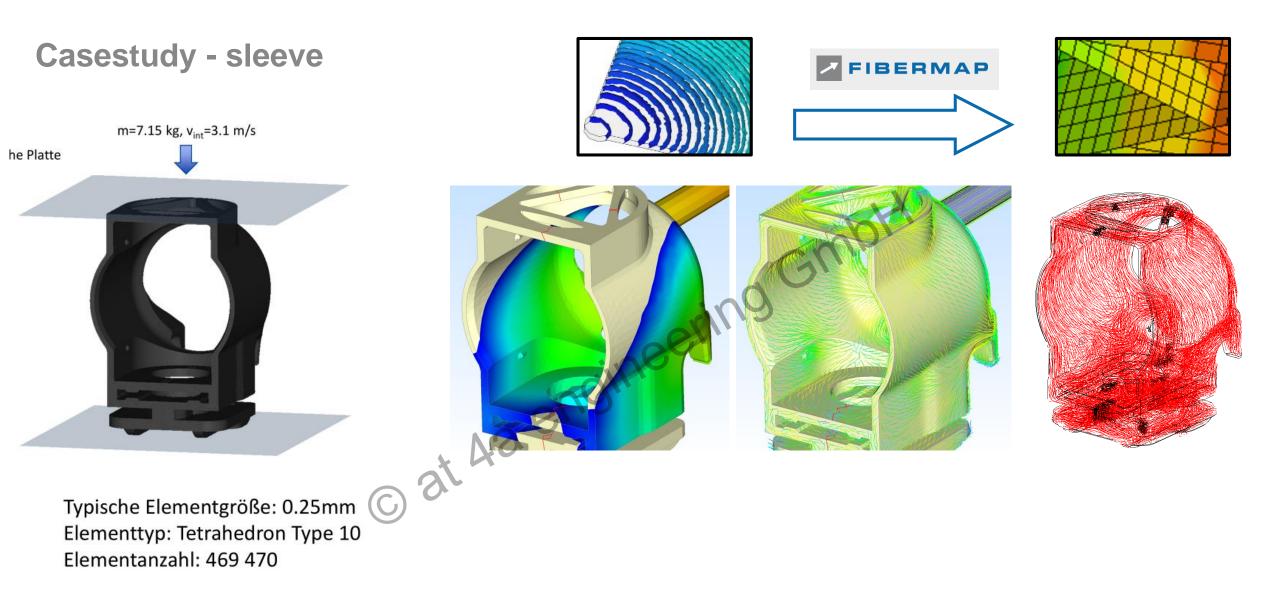
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Case Study





See more: R. Steinberger, et.al. Hirtenberger Automotive Group – Considering the Local Anisotropy of Short Fiber Reinforced Plastics, European Dynaforum 2017

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Casestudy - sleeve



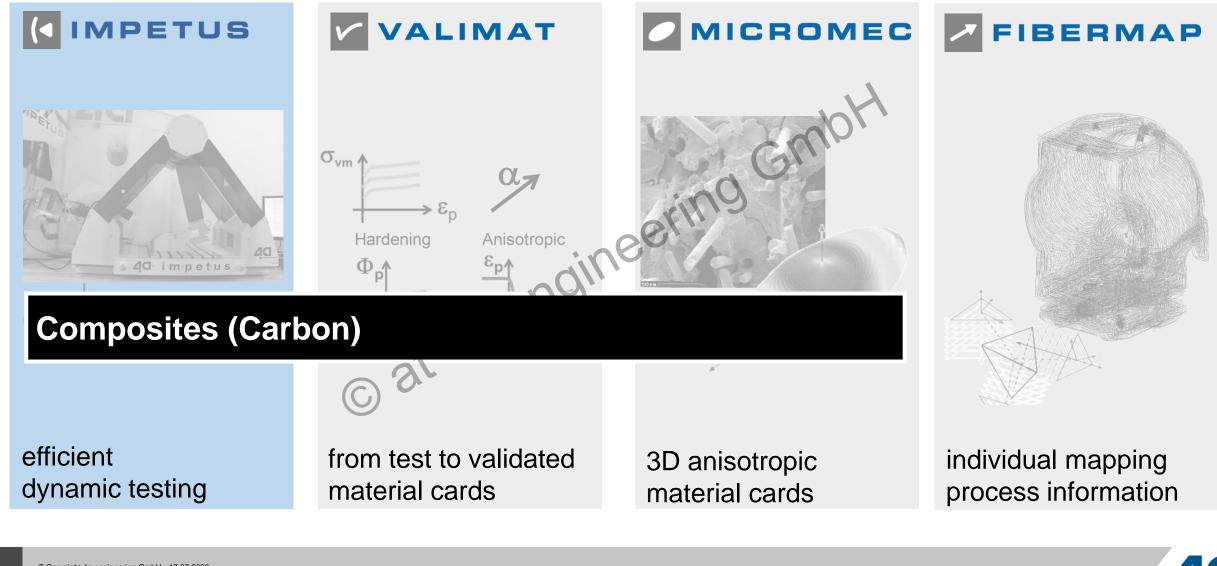
Typische Elementgröße: 0.25mm (Elementtyp: Tetrahedron Type 10 Elementanzahl: 469 470

See more: R. Steinberger, et.al. Hirtenberger Automotive Group – Considering the Local Anisotropy of Short Fiber Reinforced Plastics, European Dynaforum 2017

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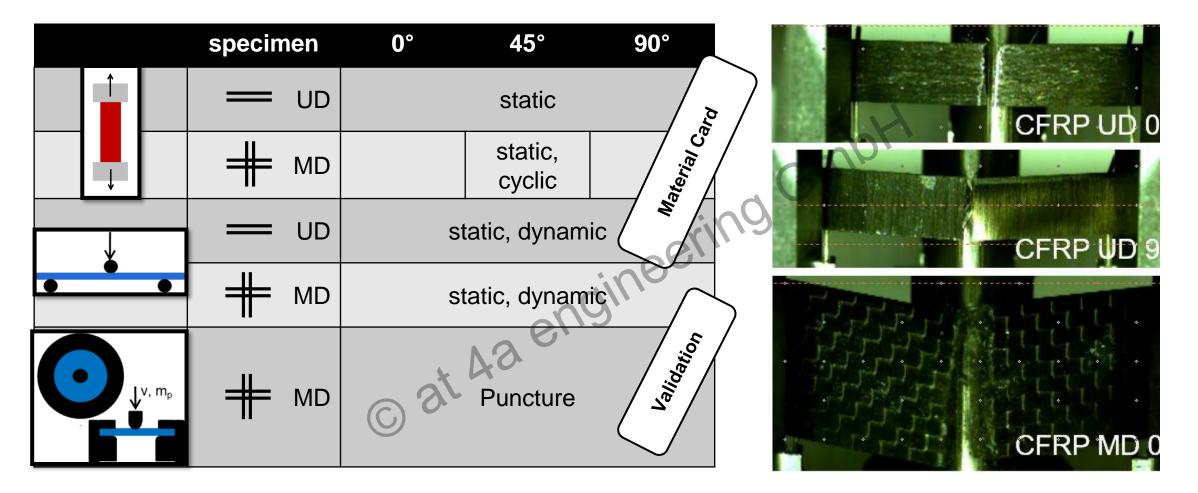
Intelligent reliable solutions for plastics, composites, metals, foams, ...



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Composites – typical test setup









| No. | Elastic | Plastic | Damage | Strain rate | Failure | |
|-------|-------------|---------|------------------------|------------------------|---|----------------------------|
| 22 | Orthotropic | None | None | None | Orientation dependent | X |
| 54/55 | Orthotropic | None | Elastic Orthotropic | Strength | Chang-Chang/Tsai-Wu Orientation dependent | lar |
| 58 | Orthotropic | None | Elastic Orthotropic | Strength, Stiffness | mod. Hashin Orientation dependent | lass, Kevlar s & fabric |
| 158 | Orthotropic | None | Elastic Orthotropic | Visco-elasticity | Orientation dependent | , G |
| 261 | Orthotropic | None | Elastic Orthotropic | None | failure Pinho (Puck) Orientation dependent | Carbon endi |
| 262 | Orthotropic | None | Elastic Orthotropic | None | failure Camanho (Puck) Orientation dependent | |
| • | • | (C | | | | |



Composites – micro mechanical approach



| | | | | es Viewer a(t) v(t) s(t) F(t) F(s) E(t) sig(ep ayers | | |
|--|--|--------------|--|---|--|--|
| | | | Model settings | | | |
| | | | Material | | | |
| *Part Com | posite | | Idealization | | | |
| — | L | | Material behaviour | | | |
| | | | Material source | Implemented | | |
| | | | | *MAT_COMPOSITE_DAMAGE (*MAT_022) | | |
| | | | Materialcardcase | 7500_MAT22 | | |
| | | | Damage/Failurecase | Chang Chang | | |
| | V | | Materialcard id | 100000 | | |
| | | | Density | 1480 | | |
| Material behaviour | | | Plasticity | None | | |
| Naterial source | Implemented | | Function (Hardening, Elastic curve form) | | | |
| Elasticity | Not isotropic elastic | | Strain rate dependency | None | | |
| Plasticity | Not selected | | Micromec | Endless fiber reinforced plastics | | |
| Failure/Damage | Not selected | | E Matrix | | | |
| Aaterial card | *MAT_ANISOTROPIC_ELASTIC_PLASTIC (*MAT_157) | - | Density of the matrix | 1093 | | |
| Deformation | *MAT_COMPOSITE_DAMAGE (*MAT_022) | | E-Modulus | 3000 | | |
| Damage/Failure | *MAT_ENHANCED_COMPOSITE_DAMAGE (*MAT_054) | | Poisson's ratio | 0.3 | | |
| Materialcard id | *MAT_LAMINATED_COMPOSITE_FABRIC (*MAT_058) | | Yield strength | 50 | | |
| Density | *MAT_RATE_SENSITIVE_COMPOSITE_FABRIC (*MAT_158) | | Strength at Break | 70 | | |
| Plasticity | *MAT_LAMINATED_FRACTURE_DAIMLER_PINHO (*MAT_261) | | Failure strain | 0.05 | | |
| | | | ⊟ Fiber | | | |
| Function (Hardening, Elastic curve for | *MAT_ANISOTROPIC_ELASTIC_PLASTIC (*MAT_157) | | Fillerlength | 20000 | | |
| Strain rate dependency | *MAT_MICROMEC (*MAT_215) | | Fillerdiameter | 20 | | |
| Micromec | *MAT_MICROMEC (*MAT_215)+Carbon | | Phi or Psi | φ | | |
| Fracture | | | Phi | 58 | | |
| Postfractura | Nana | | Psi | 71.7 | | |
| | | Г М Л | Fillermaterial | Т300 | | |
| alladie LS-DYM | VA materialcards in VALIMAT | | Orientation | | | |
| | | | Fillerorientationtype | UD | | |
| | | | ☐ Strength | | | |
| | | | Strength evaluation | Fiber strength | | |
| | | | XT | 2300 | | |

XC

Fracture

Postfracture

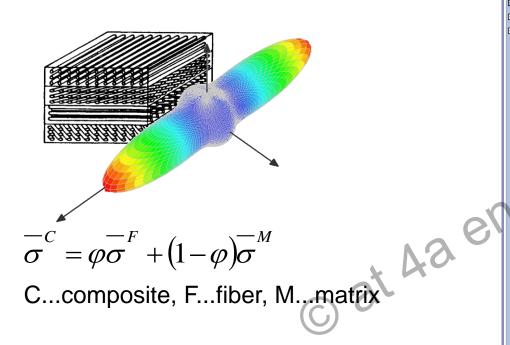
2000

None

Composite

Composites – micro mechanical approach

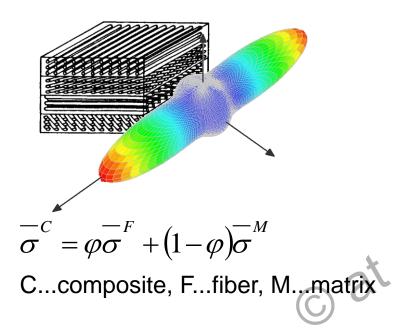




| Test Test database Measurement Report Measurement curves View | ewer a(t) v(t) s(t) F(t) | F(s) E(t) sig(eps) e | | | | |
|---|----------------------------|--------------------------------------|--|--|--|--|
| 160826_004 Material Designvariablen Layers | | | | | | |
| | | | | | | |
| Model settings | | | | | | |
| Material | | | | | | |
| Idealization | | | | | | |
| Material behaviour | | | | | | |
| Material source | Implemented | | | | | |
| Material card | *MAT_COMPOSITE_DAM | MAGE (*MAT_022) | | | | |
| Materialcardcase | 7500_MAT22 | | | | | |
| Damage/Failurecase | Chang Chang | | | | | |
| Materialcard id | 1000000 | | | | | |
| Density | 1480 | | | | | |
| Plasticity | None | | | | | |
| Function (Hardening, Elastic curve form) | | | | | | |
| Strain rate dependency | None | | | | | |
| Micromec | Endless fiber reinforced p | plastics | | | | |
| Matrix | | | | | | |
| Density of the matrix | 1093 | | | | | |
| E-Modulus | 3000 | Matrix propertie | | | | |
| Poisson's ratio | 0.3 | | | | | |
| Yield strength | 50 | Matrix properties | | | | |
| Strength at Break | 70 | | | | | |
| Failure strain | 0.05 | | | | | |
| Fiber | 00000 | | | | | |
| Fillerlength | 20000 | | | | | |
| Fillerdiameter | 20 | | | | | |
| Phi or Psi | φ | Filler properties | | | | |
| Phi Psi | 58 | | | | | |
| | 71.7 | | | | | |
| Fillermaterial | T300 | | | | | |
| Fillerorientation | | | | | | |
| Fillerorientationtype | | Orientation | | | | |
| Strength evaluation | Fiber strength | | | | | |
| XT | 2300 | Strongth | | | | |
| XC | 2000 | Strength | | | | |
| The Fracture | Composite | | | | | |
| Postfracture None | | | | | | |
| | | | | | | |

Composites – micro mechanical approach



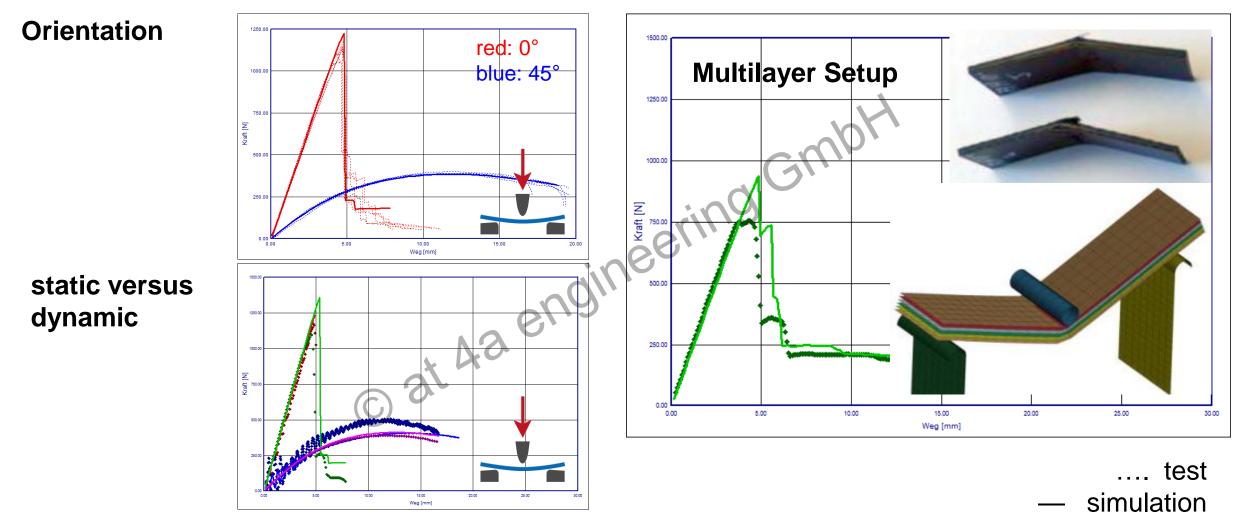


| | 160826_00 | | 326_004 | Material | | Designvariablen | | Layers | |
|--------------------------|-----------|----------|--------------|--------------|--------|-----------------|----------|---------|--------------------------------------|
| | | | | | | | | | |
| | | Name | Start | const | from | to | Variance | e Condi | Description |
| GroupName: 10_elasticity | | | | | | 0 | 3 | | |
| | | c_E11 | MMEC | | 100000 | 180000 | (NULL) | | young modulus tensile in 1 direction |
| | | c_E22 | MMEC | v . | (NHLD | (NULL) | (NULL) | | young modulus tensile in 2 direction |
| | | c_E33 | MMEC | | (NULL) | (NULL) | (NULL) | | young modulus tensile in 2 direction |
| | | c_nue21 | MMEC | | (NULL) | (NULL) | (NULL) | | poisson ration in 21 plane |
| | | c_nue31 | MMEC | ~ | (NULL) | (NULL) | (NULL) | | poisson ration in 31 plane |
| | _ | | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | poisson ration in 32 plane |
| | 2 | CG12 | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | shear modulus in 12 plane |
| X | 2 | c_G23 | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | shear modulus in 23 plane |
| | | c_G31 | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | shear modulus in 31 plane |
| | ^ | GroupNam | ie: 51_failu | re | | | | | |
| | | fc_R11T | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | |
| | | fc_R11C | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | |
| • | | fc_R22T | MMEC | v | (NULL) | (NULL) | (NULL) | | |
| | | fc_R22C | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | |
| | | fc_R12 | MMEC | \checkmark | (NULL) | (NULL) | (NULL) | | |

Typical Design Variables for *MAT_022

Composites – validation with final material card





P. Reithofer (4a engineering GmbH) & B. Fellner (MAGNA STEYR Engineering Austria) - Materialcharakterisierung von Composites; 4a Technologietag 2015

IN PHYSICS WE TRUST

Composite – case study

- Front hood
 - Stiffness versus pedestrian safety
- Material card

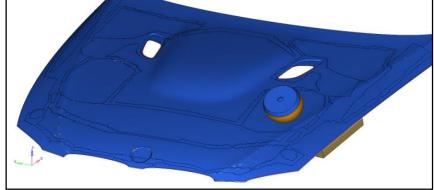
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- Composite layup with anisotropic material behavior
- Core material Honeycomb different compression levels







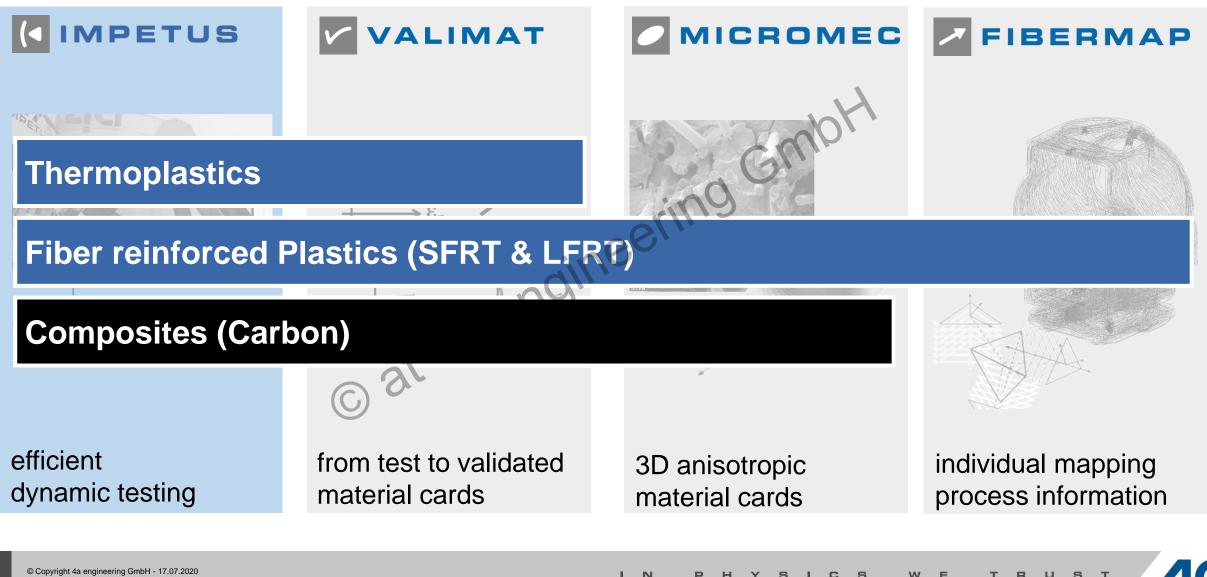


SOURCE: LINK to PAPER





intelligent reliable solutions for plastics, composites, metals, foams, ...



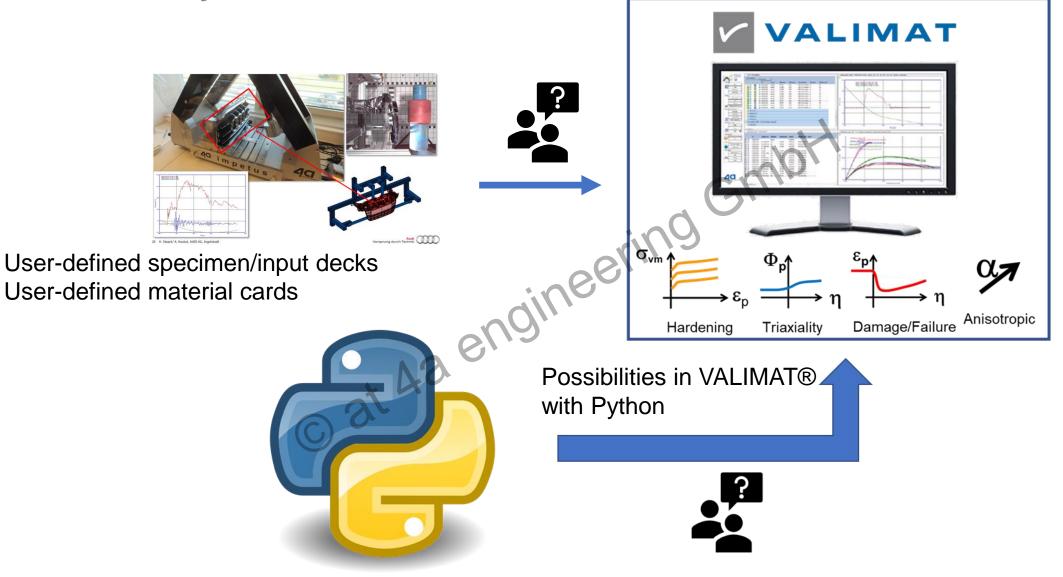
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Summary

- different models needed for different materials
 - COMPOSITE (higher fiber content) \rightarrow elastic + damage
 - SFRT, LFRT (medium fiber content) \rightarrow viscoplasticity also of importance
- Two main approaches
 - Macro scale composite behavior
 - Micro scale distinguish between fiber and resin/matrix behavior
- Differences
 - Flexibility in usage (Micro-
 - Computational effort (Macro ++)



DAY 8 – 17th July 2020



Thank you for your Attention!

4a summer-school - webinar and training Evaluating and checking test data Interpretation of typical results

SAVE THE DATE 17th Python: a powerful tool with VALIMAT®, user defined material cards/specimen

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